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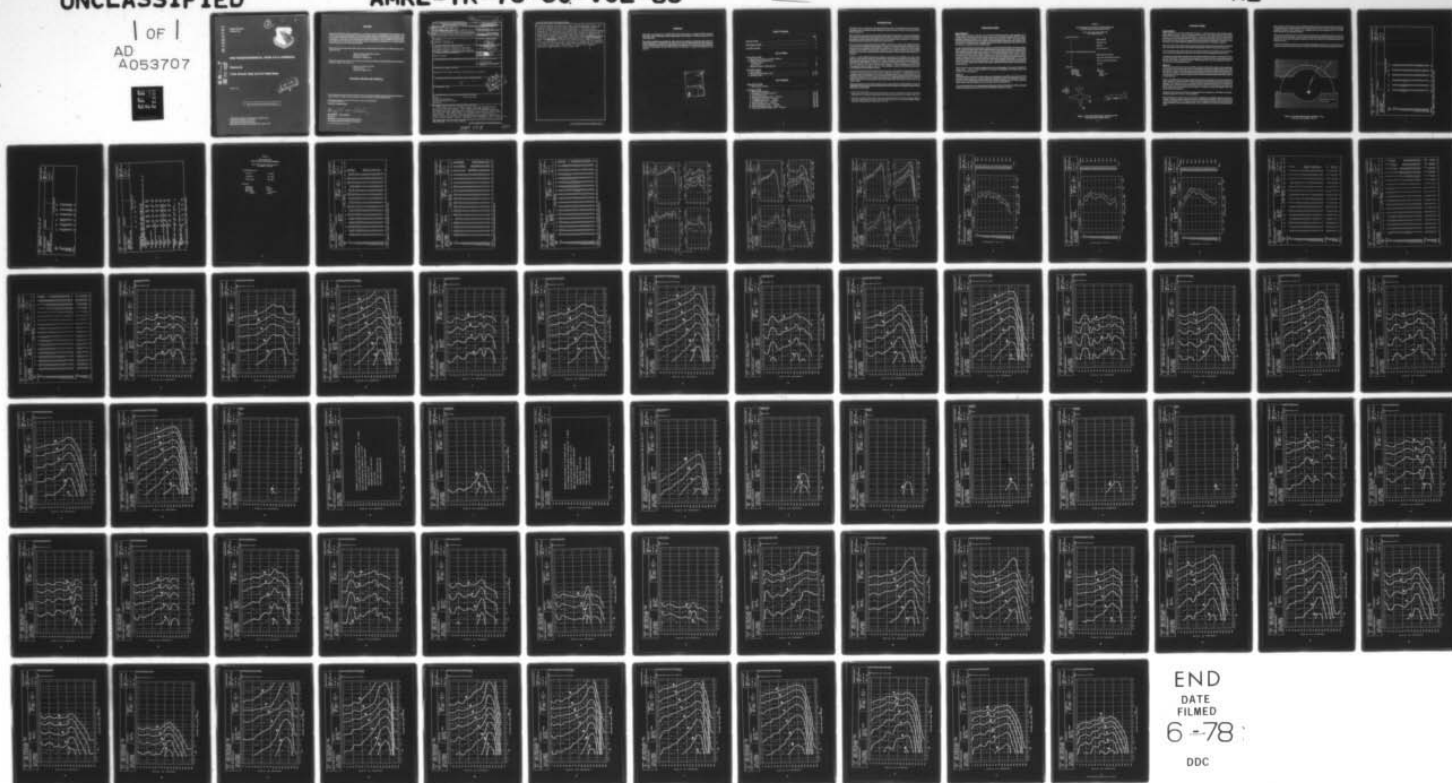
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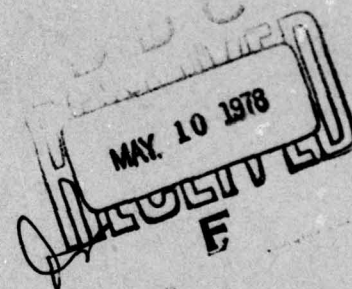


USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK

Volume 88

T-33A Aircraft, Near and Far-Field Noise

APRIL 1977



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AEROSPACE MEDICAL DIVISION
AIR FORCE SYSTEMS COMMAND
WRIGHT-PATTERSON AIR FORCE BASE, OHIO 45433

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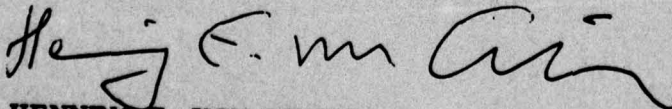
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This technical report has been reviewed and is approved for publication.

FOR THE COMMANDER



HENNING E. VON GIERKE

Director

Biodynamics and Bioengineering Division
Aerospace Medical Research Laboratory

9 Technical rept.

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preferred speech interference level, perceived noise level, and limiting times for total daily exposure of personnel with and without standard Air Force ear protectors. Far-field data measured at 19 locations are normalized to standard meteorological conditions and extrapolated from 75-8000 meters to derive sets of equal-value contours for these same seven acoustic measures as functions of angle and distance from the source. Refer to Volume 1 of this handbook, "USAF Bioenvironmental Noise Data Handbook, Vol 1: Organization, Content and Application", AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc.

PREFACE

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 723104, Measurement and Prediction of Noise Environments of Air Force Operations.

The authors gratefully acknowledge Mr. John Cole for his assistance in preparing this report, Mr. Robert England for his assistance in acquiring the raw data, Mr. Keith Kettler, Mr. Henry Mohlman and Mr. David Eilerman of the University of Dayton for assistance in the mechanics of data processing, and Ms. Norma Peachey and Mr. Mike Patterson for assistance in typing and preparation of the graphics.

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INTRODUCTION

The USAF T-33A is a flight trainer aircraft powered by one J33-A-35 turbojet engine. The aircraft was manufactured by the Lockheed Aircraft Corporation and the engines by Allison, a Division of General Motors Corporation.

This volume provides measured and extrapolated data defining bioacoustic environments produced by this aircraft during ground runup operations. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with ground runups of the T-33A aircraft.

This volume is one of a series published by the AMRL under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and ground support equipment. The far-field, community-type, noise data in the handbook describe the noise produced during *ground operations* of aircraft, ground support equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. Volume 2 provides a method and data for adjusting the handbook's far-field noise data, which are for standard meteorological conditions (15 C temperature, 70% rel humidity, 0.760 meters Hg barometric pressure), to derive comparable data for other meteorological conditions. *Refer to Volumes 1 and 2* (references 1 and 2) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., inflight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published and is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of each updated index.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; AUTOVON 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

1. Cole, John N., *USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application*, AMRL-TR-75-50 (1) Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.
2. Cole, John N., *USAF Bioenvironmental Noise Data Handbook, Volume 2: Procedure to Evaluate Effects of Non-standard Meteorological Conditions on Far-Field Noise*, AMRL-TR-75-50 (2), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

NEAR-FIELD NOISE

MEASUREMENTS

AMRL acquired near-field noise data on the T-33A aircraft during ground runup operations of its turbojet engine. For these tests the aircraft was located on the "Hot Cargo" pad, Eglin AFB, FL, with no significant reflecting surfaces in the vicinity except the ground plane. Table 1 gives the surface meteorological conditions and the four engines and ground support equipment power conditions. The ground-crew chief selected power conditions and near-field locations generally used during routine maintenance or engine runup for preflight checks.

At each near-field location a test engineer randomly moved a hand-held microphone in and around each location, probing all areas where a crew member's head would normally be located. He recorded all the noise samples on magnetic tape. During analysis of each sample he determined the one-third octave band root-mean-square sound pressure using a 4- or 8-second integration time to derive a power-averaged level for each location. Figure 1 shows the four near-field locations where ground crew are usually located for maintenance and/or preflight checkout operations. Estimates of noise levels at other locations are difficult in the near-field since the noise source is spatially distributed, i.e., not a point source. The noise levels at near-field locations can vary widely depending upon relative distances from each noise source (intake noise, exhaust noise, panel resonances, internal engine noise through the engine wall, etc.).

Table 1 lists the numeric/alphabetic designators used on the data pages in this report to identify the measurement locations and test conditions. For example, the designator 1/A means ground crew location 1 and test condition A.

RESULTS

The measured data presented in Table 2 define the sound pressure levels (SPL) produced by the T-33A aircraft at the four ground crew locations. This table includes the overall, 1/3 octave band, and octave band levels. From these data one can calculate the variety of measures given in Table 3, which are widely used to assess the effects of noise on personnel and their performance.

All near-field data are for the meteorological conditions at the time of test but are valid for all typical airbase meteorology because of the short sound propagation distances involved.

TABLE 1

MEASUREMENT LOCATIONS AND TEST CONDITIONS
FOR NEAR-FIELD NOISE MEASUREMENTS

T-33A Aircraft, Ground Runup, Eglin AFB
Tail # 63655, 15 July 1971

Ground Crew Location

1	Operator MD-3M
2	Wheel Chock Pull
3	Marshall
4	Crew Chief Observer

Aircraft Engine (and Support Equipment) Operation

A	MD-3M Operating (unloaded)
B	Engine Start, MD-3M Operating (loaded)
C	Idle Power and MD-3M (unloaded)
D	Idle Power

Meteorology

Temperature	25.6 C
Bar Pressure	0.758 M Hg
Rel Humidity	85 %
Wind — Speed	1 M/Sec (2 Kt)
— Direction	270 Deg.

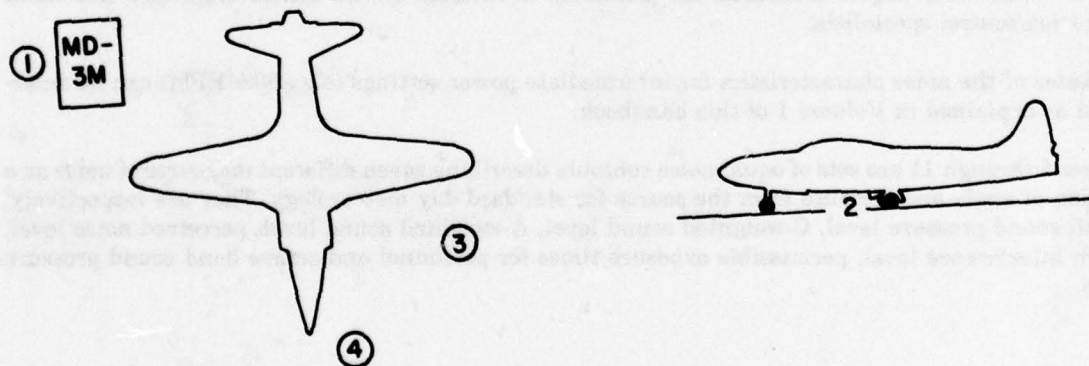


Figure 1. Near-Field Measurement Locations on the
Hot Cargo Pad at Eglin AFB FL

FAR-FIELD NOISE

MEASUREMENTS

AMRL acquired the near and far-field data during a 1- 2-hour test period, thus keeping similar meteorological conditions. Figure 2 shows the aircraft on the "Hot Cargo" pad and its orientation relative to 19 microphone measurement sites on a semicircle. The center of the 75 meter radius semicircle used in surveying the J33-A-35 engine was on the ground directly below the intersection of the aircraft's centerline and the plane passing through the engine's exhaust plane. The ground runup pad did not have a blast deflector; therefore, the jet exhaust was in a "free-flow" condition.

Table 4 provides cockpit readouts of engine speed in percent for each power setting used in the far-field tests. Also listed in this table are the surface meteorological conditions during data acquisition.

All 19 microphone measurement sites are in the acoustic far-field of their respective source where the sound wave-fronts spherically diverge and the noise source may be regarded as a point source.

A portable microphone/tape recorder system was used to sequentially record 5 to 10 seconds of noise at each far-field location. The microphone was hand-held 1.7 meters (5-1/2 feet) above the ground and pointed at the source (0° angle of incidence). These samples were then time-integrated to derive a root-mean square sound pressure level.

RESULTS

Table 5 lists the overall and 1/3 octave band SPL measured at the far-field locations under meteorological conditions at the time of the test. Data in all other figures and tables are based on these levels. These data were normalized to 100 meters distance and standard meteorological conditions (15 C temperature, 70% relative humidity, 0.760 meter Hg barometric pressure) and used to derive the graphic data in Figure 3 which provides a compact summary of the far-field noise characteristics of the T-33A aircraft in a standard format.

Figure 4 and Table 6 present two basic acoustic measures, the acoustic power levels and the directivity index, respectively. The acoustic power level describes the power radiated by the source as a function of frequency. The directivity index is a standard acoustical engineering measure that describes the geometric way in which the source radiates this power as a function of both frequency and angle from source. These basic source measures are primarily of interest for acoustical engineers and noise generation/control specialists.

Estimates of the noise characteristics for intermediate power settings (e.g., 80% RPM) can be determined as explained in Volume 1 of this handbook.

Figures 5 through 11 are sets of equal noise contours describing seven different measures of noise as a function of angle and distance from the source for standard day meteorology. They are respectively, overall sound pressure level, C-weighted sound level, A-weighted sound level, perceived noise level, speech interference level, permissible exposure times for personnel and octave band sound pressure levels.

Data excessively influenced by spurious background/electronic noise were eliminated from all figures and tables. No data are presented beyond the 160 degree location for the idle power settings because of background/electronic noise. Typically, the A-weighted levels for these angles are from 5 to 10 dBA below the level at the 160 degree location.

Test personnel performed noise surveys during quiet periods when the background noise was minimal, e.g., early in the morning when no other aircraft or engine test stands were operating. Data eliminated because they were near the background/electronic noise were generally not significant because the levels were so low.

Volume 2 of the handbook describes the influence of meteorology on far-field noise environments, and provides, if required, the factors necessary to adjust the handbook's standard meteorological day data.

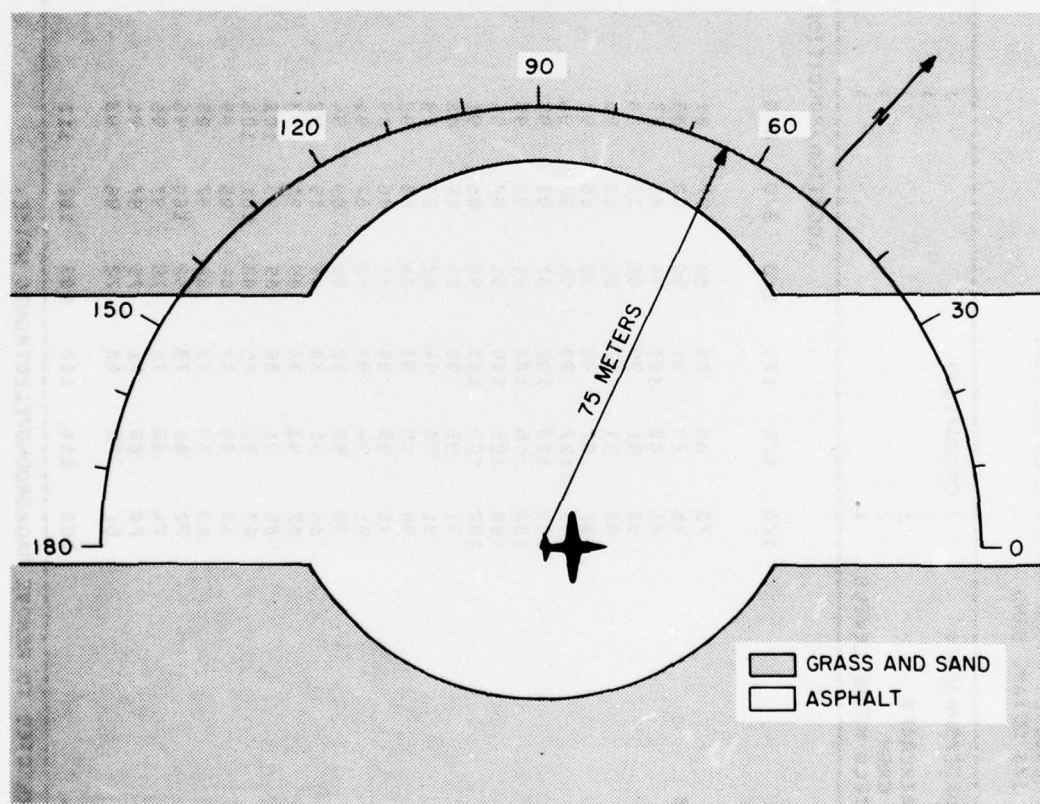


Figure 2. Far-Field Measurement Locations on the Hot Cargo Pad at Eglin AFB, FL

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)							IDENTIFICATION:	
2 1/3 OCTAVE BAND							OMEGA 3.2	
							TEST 71-019-100	
NOISE SOURCE/SUBJECT:							RUN 01	
(OPERATION:								
T-33A AIRCRAFT								
GROUND CREW							04 DEC 74	
NEAR FIELD NOISE LEVELS							PAGE F1	
							LOCATION/CONDITION	
FREQ (HZ)	1/A	1/B	1/C	2/D	3/D	4/D		
25	73	69	85	83	77	81		
31.5	76	72	91	90	80	85		
40	81	80	101	94	87	88		
50	93	91	99	88	82	88		
63	93	91	97	86	83	90		
80	87	87	92	85	91	91		
100	98	107	99	93	93	96		
125	100	108	102	87	91	97		
160	100	106	108	90	90	95		
200	100	105	105	91	87	91		
250	102	106	103	90	86	91		
315	91	96	93	86	85	88		
400	91	98	94	82	86	89		
500	90	96	93	84	88	90		
630	90	95	94	86	86	89		
800	91	94	92	83	87	91		
1000	92	93	92	88	89	94		
1250	86	91	88	89	92	92		
1600	87	93	88	85	90	93		
2000	85	91	86	85	95	100		
2500	83	91	85	88	101	106		
3150	80	89	82	86	95	95		
4000	80	89	82	86	97	95		
5000	77	86	79	82	100	96		
6300	74	83	76	79	97	96		
8000	70	80	71	75	97	94		
10000	67	75	68	72	96	93		
OVERALL	108	114	112	103	107	110		

TABLE: MEASURED SOUND PRESSURE LEVEL (08)		IDENTIFICATION:				
2	OCTAVE BAND					
NOISE SOURCE/SUBJECT:		OPERATION:				
T-33A AIRCRAFT	((OMEGA 3.2			
GROUND CREW	((TEST 71-019-100			
NEAR FIELD NOISE LEVELS	((RUN 01			
	((04 DEC 74			
	((PAGE J1			
		LOCATION/CONDITION				
FREQ (HZ)	1/A	1/B	1/C	2/D	3/D	4/D
31.5	83	81	101	95	85	89
63	96	95	102	94	88	94
125	104	112	109	95	96	100
250	104	109	107	94	91	95
500	95	101	98	89	91	94
1000	95	97	96	96	93	97
2000	90	96	91	91	102	107
4000	84	93	86	89	102	100
8000	76	85	77	81	101	99
OVERALL	108	114	112	103	107	110

TABLE: MEASURES OF HUMAN NOISE EXPOSURE							IDENTIFICATION:
3							OMEGA 3.2
							TEST 71-819-108
NOISE SOURCE/SUBJECT: (OPERATION:)							RUN 81
T-33A AIRCRAFT ()							
GROUND CREW ()							04 DEC 74
NEAR FIELD NOISE LEVELS ()							PAGE W1
							</

TABLE 4
TEST CONDITIONS
FOR FAR-FIELD NOISE MEASUREMENTS

T-33A Aircraft Ground Runups, Eglin AFB, FL
Tail # 63655, 15 July 1971

Aircraft Engine Operation

Idle Power	35 % RPM
Runup Power	50 % RPM
Military Power	100 % RPM

Meteorology

Temperature	25.6 C
Bar Pressure	0.758 M Hg
Rel Humidity	85 %
Wind — Speed	1 M/Sec (2 Kts)
— Direction	270 Deg

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)															IDENTIFICATION:				
5 1/3 OCTAVE BAND																			
DISTANCE = 75 METERS															OMEGA 1.4				
															TEST 75-002-045				
NOISE SOURCE/SUBJECT:															RUN 03				
(OPERATION:)																			
(MILITARY POWER)																			
(100% RPM)															09 MAY 75				
(FREE FLOW)															PAGE 2				
T-33A AIRCRAFT																			
J33-A-35 ENGINE																			
FAR FIELD NOISE																			
METEOROLOGY: = 26 C																			
BAR PRESS = .758 M HG																			
REL HUMID = 85 %																			
FREQ (HZ)															ANGLE (DEGREES)				

(FIGURE: NORMALIZED FARFIELD NOISE LEVELS
 (3 DISTANCE = 100 METERS
 (NOISE SOURCE/SUBJECT: (OPERATION:
 (T-33A AIRCRAFT (IDLE POWER
 (J33-A-35 ENGINE (352 RPM
 (FAR FIELD NOISE (FREE FLOW
 (1 = 31.5 HZ 2 = 63 HZ 3 = 125 HZ
 (4 = 250 HZ 5 = 500 HZ 6 = 1000 HZ
 (METEOROLOGICAL DATA: (TEMPERATURE = 15 C
 (BAR PRESSURE = .760 M HG
 (REL HUMIDITY = 70 %
 (PAGE 6
 (IDENTIFICATION: (OMEGA 1.4
 (TEST 75-002-045
 (RUN 01
 (09 MAY 75
 ()

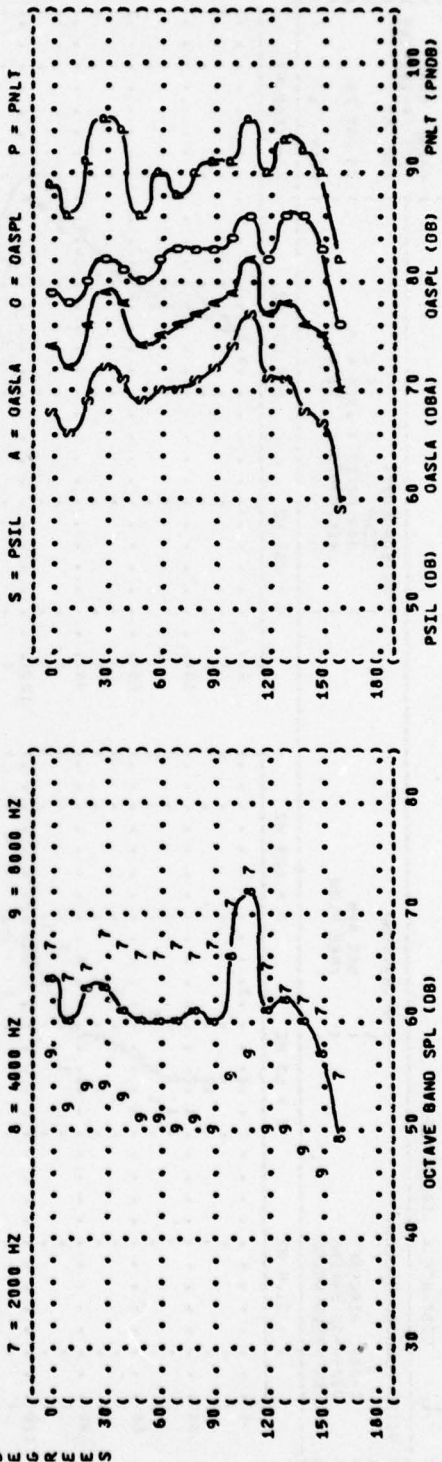
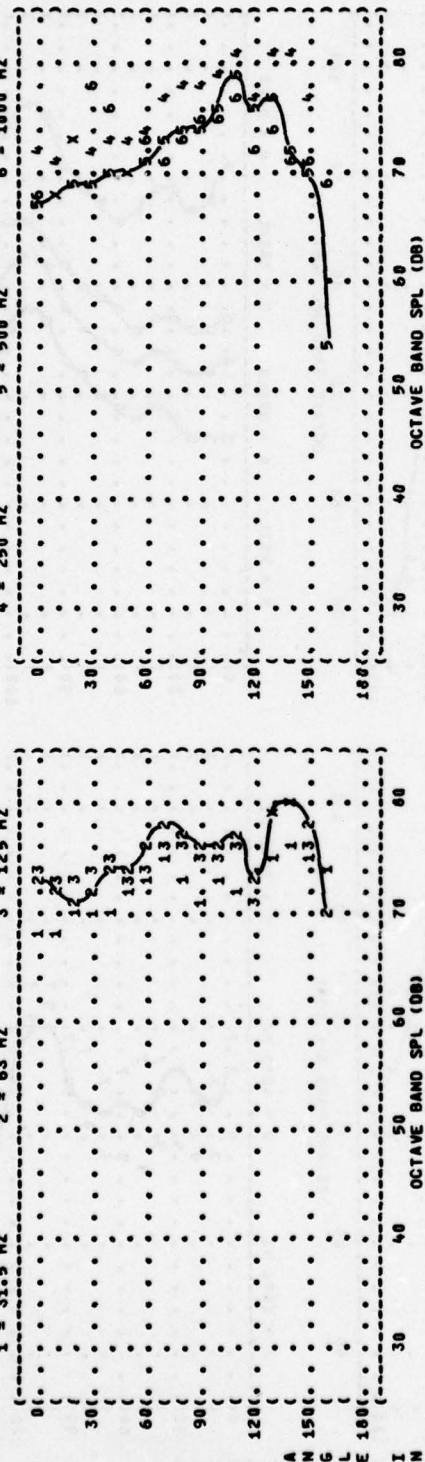


FIGURE: NORMALIZED FARFIELD NOISE LEVELS

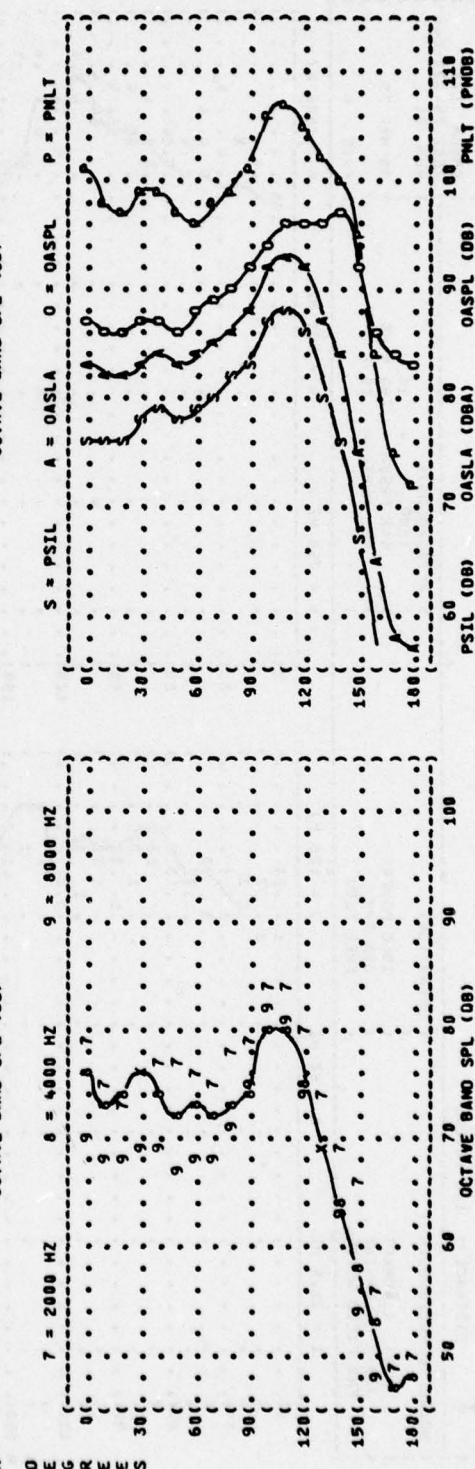
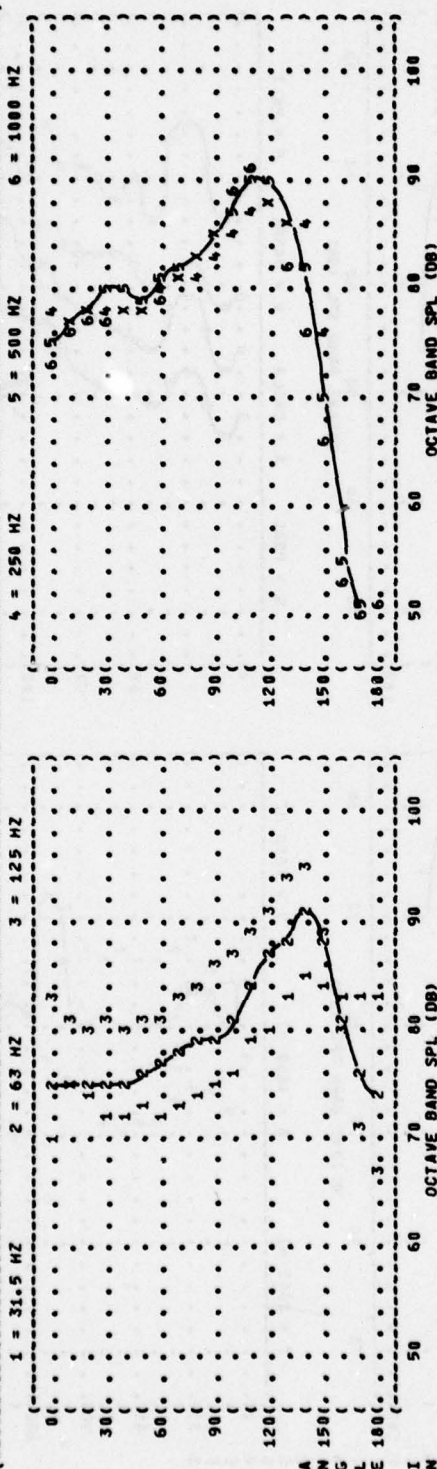
3 DISTANCE = 100 METERS

NOISE SOURCE/SUBJECT: T-33A AIRCRAFT
J33-A-35 ENGINE
FAR FIELD NOISE

OPERATION: 50% RPM
FREE FLOW

METEOROLOGY: TEMP = 15 C
BAR PRESS = .760 Hg
REL HUMID = 78 %

IDENTIFICATION: OMEGA 1.4
TEST 75-082-045
RUN 02
89 MAY 75
PAGE 6



(FIGURE: NORMALIZED FARFIELD NOISE LEVELS
 (3 DISTANCE = 100 METERS
 (NOISE SOURCE/SUBJECT: (OPERATION:
 (T-33A AIRCRAFT (MILITARY POWER
 (J33-A-35 ENGINE (1002 RPM
 (FAR FIELD NOISE (FREE FLOW
 (METEOROLOGY: (TEMPERATURE = 15 C
 (BAR PRESS = .760 M HG
 (REL HUMID = 70 %
 (PAGE 6
 (IDENTIFICATION:
 (OMEGA 1.4
 (TEST 75-002-045
 (RUN 03
 (09 MAY 75
 ()

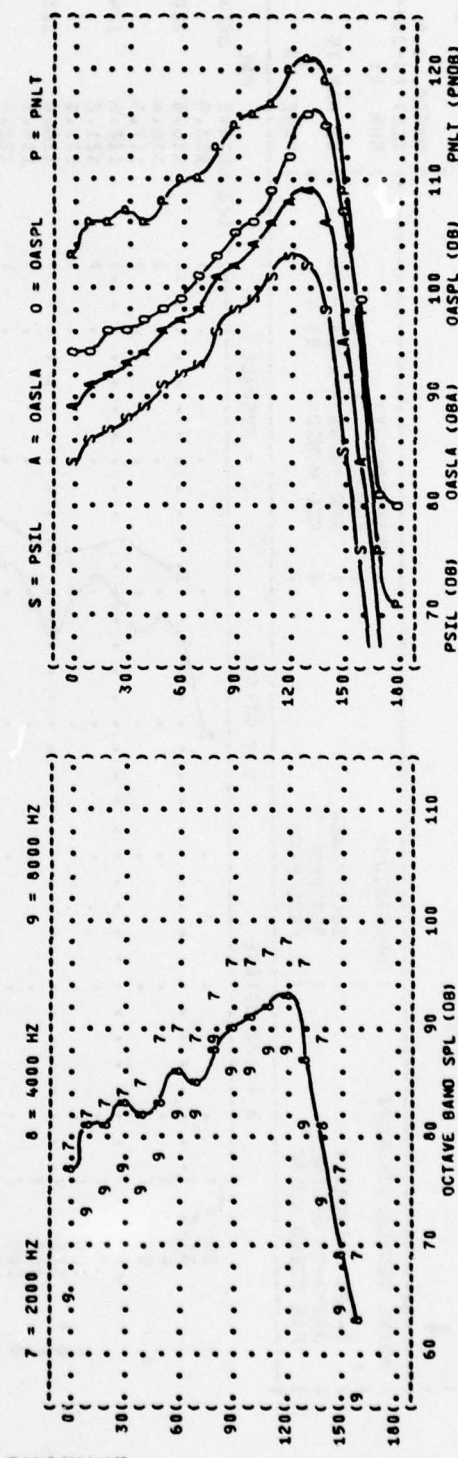
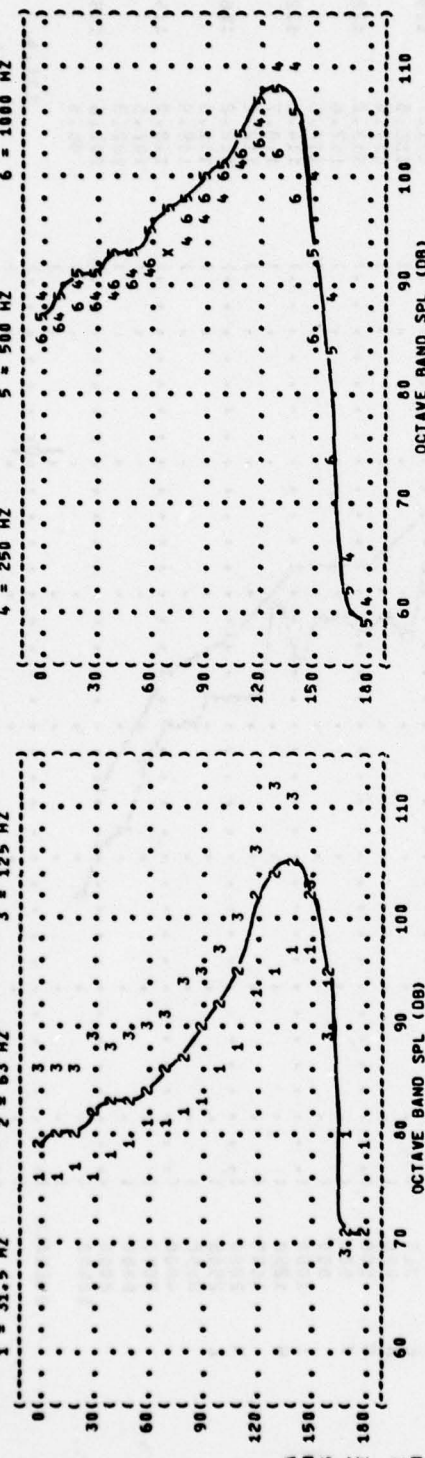
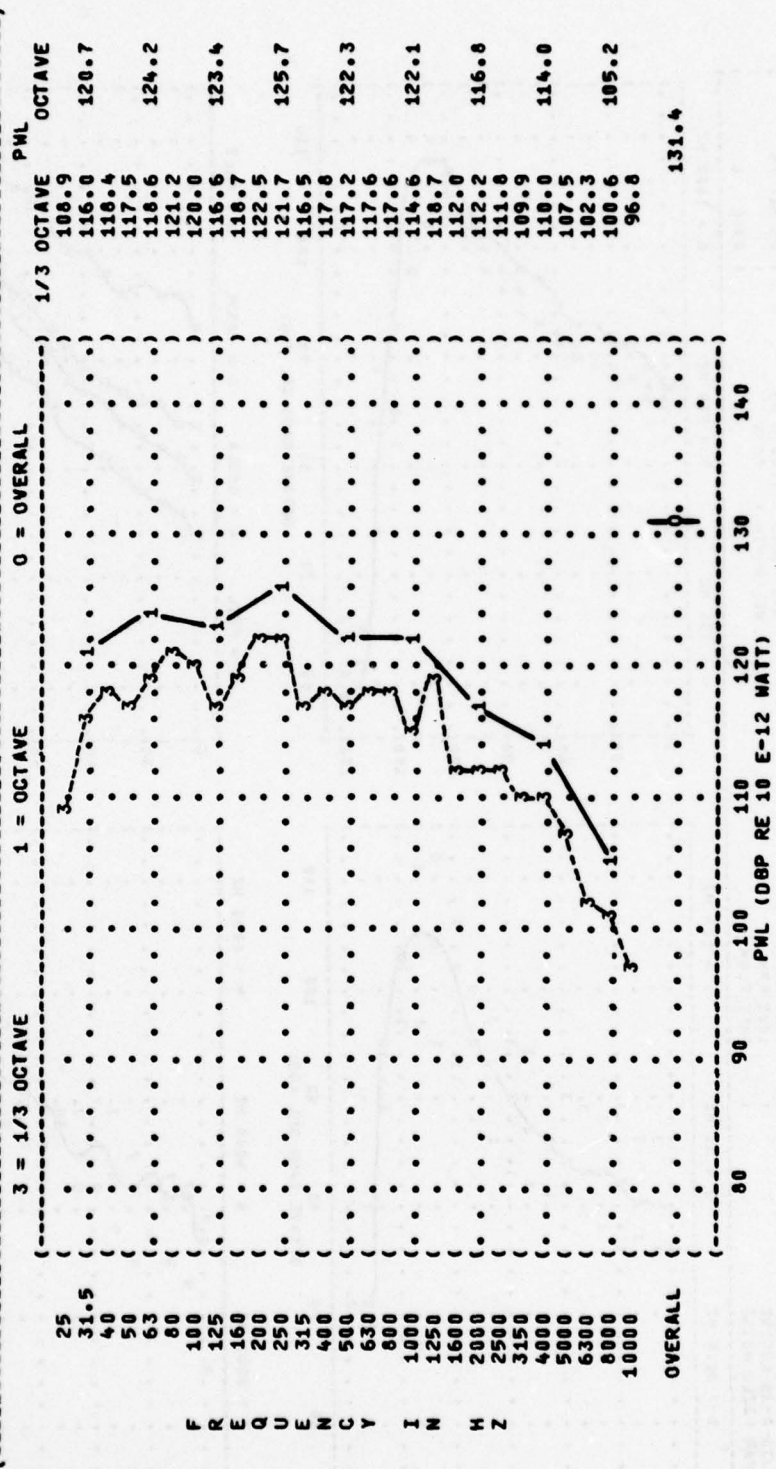


FIGURE 1. ACOUSTIC POWER LEVEL {PWL}

4

NOISE SOURCE/SUBJECT:	OPERATION:	METEOROLOGY:	TIME:
I-33A AIRCRAFT	IDLE POWER	TEMP = 26 C	RUN 01
J33-A-35 ENGINE	35% RPM	BAR PRESS = .758 M HG	09 MAY 75
FAR FIELD NOISE	FREE FLOW	REL HUMID = 85 %	PAGE 3



FREQ	PWL (DBP RE 10 E-12 WATT)		
	3 = 1/3 OCTAVE	1 = OCTAVE	0 = OVERALL
25	119.4	126.9	126.9
31.5	120.7	124.5	124.5
40	124.6	126.0	126.0
50	129.0	133.4	131.7
63	131.1	128.3	136.2
80	126.5	127.6	131.5
100	127.9	128.5	133.1
125	128.6	130.2	132.9
160	124.5	122.2	127.7
200	121.5	119.0	125.4
250	120.8	121.7	128.2
315	123.3	126.2	115.4
400	128.2	115.4	141.2
500	128.2	115.4	141.2
630	128.2	115.4	141.2
800	128.2	115.4	141.2
1000	128.2	115.4	141.2
1250	128.2	115.4	141.2
1600	128.2	115.4	141.2
2000	128.2	115.4	141.2
2500	128.2	115.4	141.2
3150	128.2	115.4	141.2
4000	128.2	115.4	141.2
5000	128.2	115.4	141.2
6300	128.2	115.4	141.2
8000	128.2	115.4	141.2
10000	128.2	115.4	141.2
OVERALL	128.2	115.4	141.2

OVERALL

19

TABLE: DIRECTIVITY INDEX (DB)																
6																
IDENTIFICATION:																
OMEGA 1.4																
TEST 75-002-045																
RUN 01																
NOISE SOURCE/SUBJECT:																
T-33A AIRCRAFT																
J33-A-35 ENGINE																
FAR FIELD NOISE																
OPERATION:																
IDLE POWER																
35% RPM																
FREE FLOW																
METEOROLOGY:																
TEMP = 26 C																
BAR PRESS = .758 M HG																
REL HUMID = 85 %																
PAGE 4																
FREQ (HZ)																
0 10 20 30 40 50 60 70 80 90 100 110 120 130 140 150 160 170 180																
1/3 OCTAVE																
25	-4	-4	-4	-2	-2	3	5	0	-0	3	-1	-1	4	4	2	4
31.5	-5	-6	-5	-4	-2	-2	3	-0	-2	1	1	-0	3	1	2	1
40	-7	-7	-6	-3	-2	-2	3	-0	-1	-1	-1	1	3	4	3	0
50	-5	-5	-7	-6	-3	-3	0	0	0	-1	0	-5	2	4	3	-4
63	-2	-3	-3	-2	-2	-2	1	0	1	-1	1	-4	3	5	2	-6
80	-1	-2	-3	-2	-2	-2	0	1	0	-1	1	-3	3	4	1	-7
100	-1	-2	-2	-2	-2	-2	-1	0	0	-1	1	-4	3	3	-1	
125	-4	-4	-3	-3	-3	-3	-1	-1	-2	-1	-0	-4	4	4	1	
160	-6	-7	-6	-7	-7	-6	-5	-4	-4	1	1	-1	3	4	0	
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500	-8	-6	-5	-4	-4	-4	-3	0	-1	1	5	3	2	-3	-3	-20
630	-9	-7	-5	-6	-4	-4	-4	-1	-1	1	6	3	2	-3	-6	-21
800	-6	-6	-3	-4	-4	-4	-4	-2	-2	2	4	-1	1	-3	-7	-19
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10000	6	2	3	2	1	0	1	-1	-3	2	3	-3	-3	-5	-6	
OCTAVE																
31.5	-5	-5	-3	-3	-3	-1	-0	2	1	-1	-0	-1	2	4	2	1
63	-3	-4	-5	-5	-3	-2	-0	1	0	-1	-1	-4	3	4	2	-6
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4000	0	-4	-1	-1	-2	-4	-4	-4	-3	2	8	-2	-4	-4	-7	-15
8000	5	0	2	2	1	-1	-1	-2	-1	3	4	-3	-3	-4	-6	
OVERALL																
	-4	-5	-4	-2	-2	-3	-2	0	0	-0	1	3	-2	3	3	-7

TABLE: DIRECTIVITY INDEX (DB)																			
IDENTIFICATION:																			
6																			
NOISE SOURCE/SUBJECT: () OPERATION: () METEOROLOGY: ()																			
T-33A AIRCRAFT () 50% RPM () TEMP = 26 C ()																			
J33-A-35 ENGINE () FREE FLOW () BAR PRESS = .750 M HG ()																			
FAR FIELD NOISE () REL HUMID = 85 % ()																			
PAGE 4																			
FREQ (HZ)	0	10	20	30	40	50	60	70	80	90	100	110	120	130	140	150	160	170	180
ANGLE (DEGREES)																			
1/3 OCTAVE																			
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31.5	-7	-3	-5	-7	-6	-7	-6	-5	-5	-3	-4	-1	0	4	5	5	6	5	4
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8000	-8	-9	-9	-8	-7	-10	-8	-10	-5	-1	7	6	0	-4	-11	-22	-28	-33	-33
10000	-0	-1	-2	0	-1	-1	-3	-5	-4	-2	5	5	1	-3	-7	-16	-21	-27	-28
OCTAVE																			
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8000	-5	-7	-7	-6	-6	-8	-7	-8	-4	0	7	6	-0	-5	-12	-21	-27	-32	-32
OVERALL																			
	-6	-7	-7	-6	-6	-7	-5	-4	-3	-1	2	3	3	3	4	-1	-7	-9	-10

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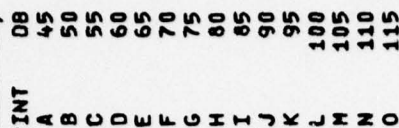
OMEGA 1.4

► METEOROLOGY:

TEMP = 15 C
BAR PRESS = .760 M HG

REL HUMID = 70 %

PAGE 13



C-WEIGHTED OVERALL SOUND LEVEL {OASLC}
EQUAL LEVEL CONTOURS (DBC)

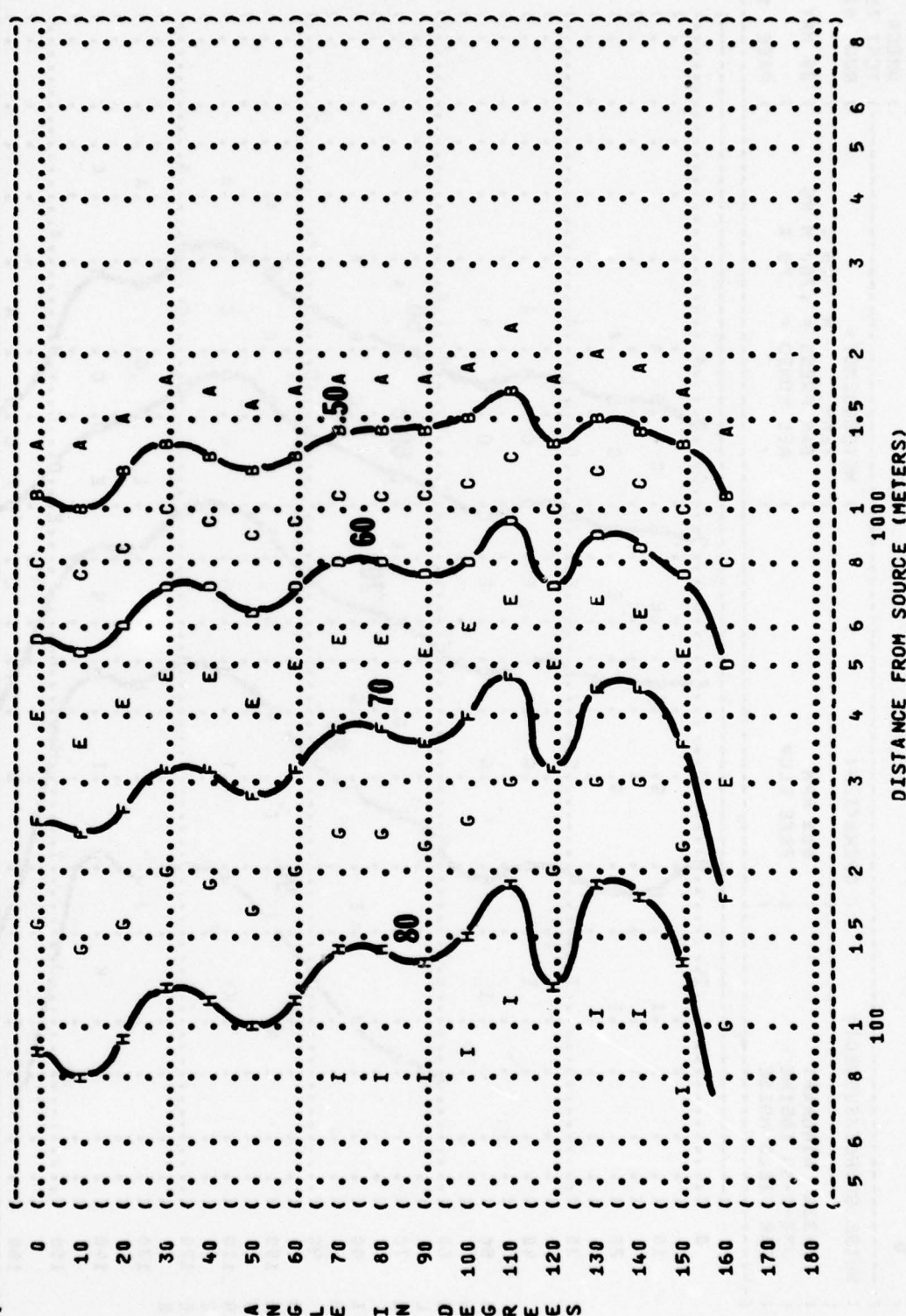
IDENTIFICATION:
OMEGA 1.4
TEST 75-002-74
RUN 01
09 MAY 75
PAGE 14

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %

**ERATION:
IDLE POWER
35% RPM
FREE FLOW**

T-33A AIRCRAFT
J33-A-35 ENGINE
FAR FIELD NOISE

POINT	DBC
A	45
B	50
C	55
D	60
E	65
F	70
G	75
H	80
I	85



ANGIE IN DEGREES

IDENTIFICATION:

OMEGA 1-4

1) METEOROLOGY:

RUN 02

BAR PRESS = .760 M HG

REL HUMID = 70 %

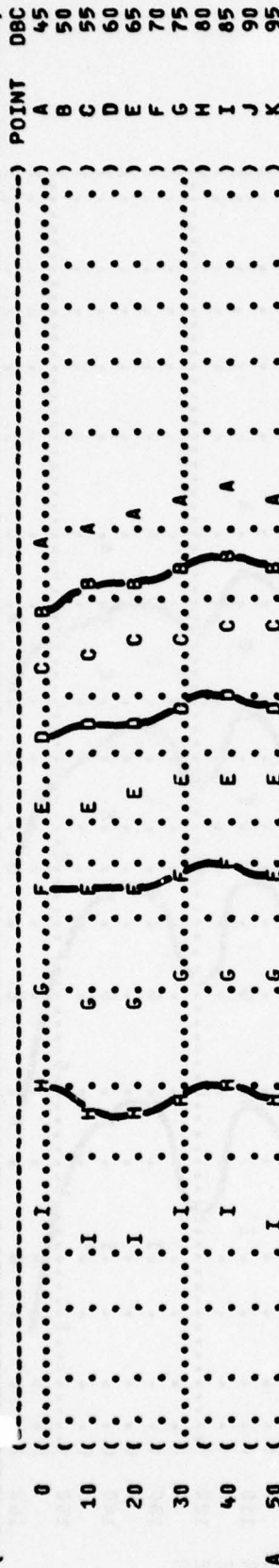
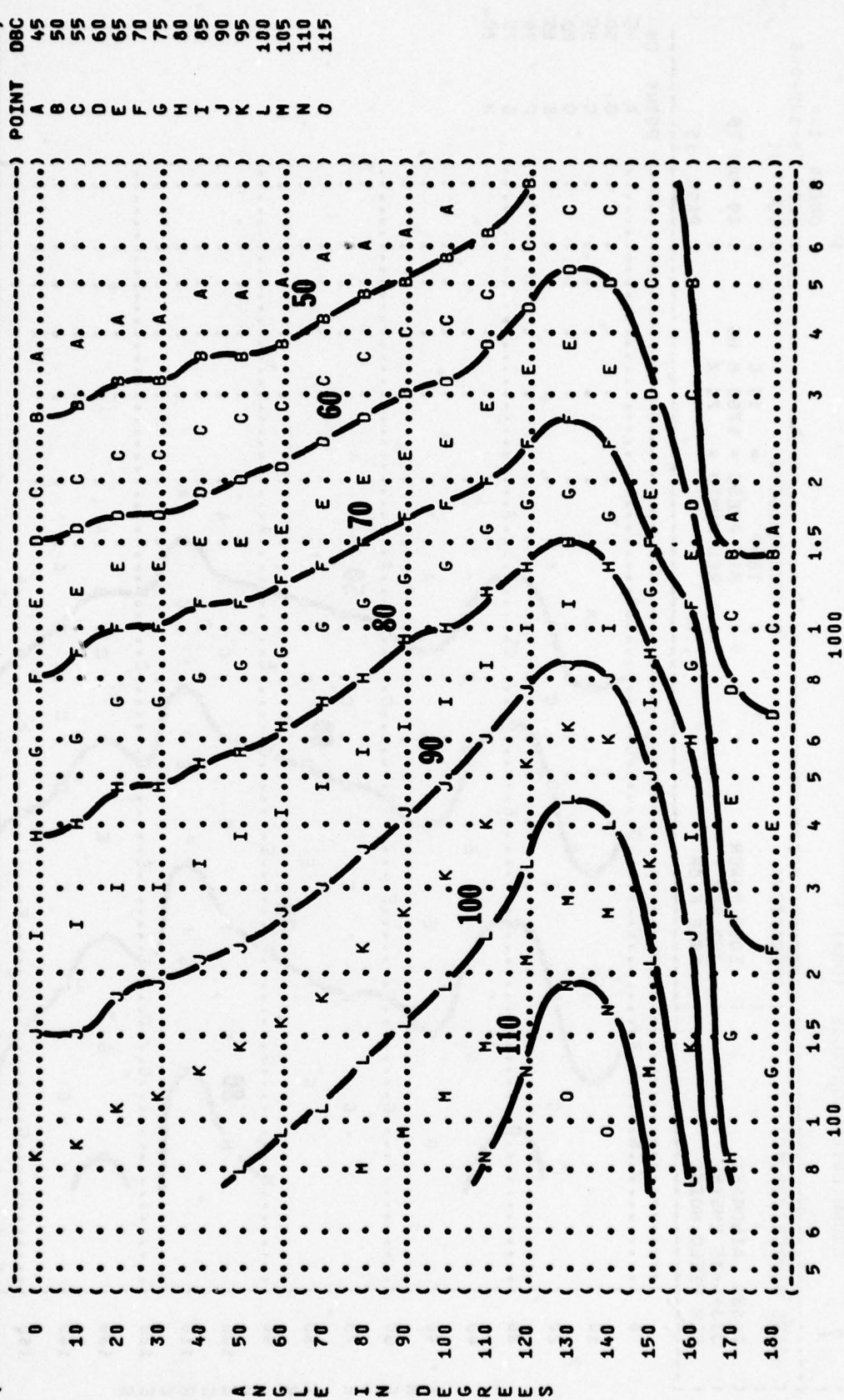
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FIGURE: C-WEIGHTED OVERALL SOUND LEVEL (OASLC)
 6
 IDENTIFICATION:
 OMEGA 1.4
 TEST 75-002-045
 RUN 03
 09 MAY 75
 PAGE 14
 METEOROLOGY:
 TEMP = 15 C
 BAR PRESS = .760 M HG
 REL HUMID = 70 %
 OPERATION:
 MILITARY POWER
 100% RPM
 FREE FLOW
 NOISE SOURCE/SUBJECT:
 T-33A AIRCRAFT
 J33-A-35 ENGINE
 FAR FIELD NOISE



DISTANCE FROM SOURCE (METERS)

POINT DBC
 A 45
 B 50
 C 55
 D 60
 E 65
 F 70
 G 75
 H 80
 I 85
 J 90
 K 95
 L 100
 M 105
 N 110
 O 115

FIGURE: A-WEIGHTED OVERALL SOUND LEVEL (OASLA)
EQUA LEVEL CONTOURS (DBA)

7

IDENTIFICATION:

OMEGA 1.4
TEST 75-002-045
RUN 01

NOISE SOURCE/SUBJECT:

OPERATION:

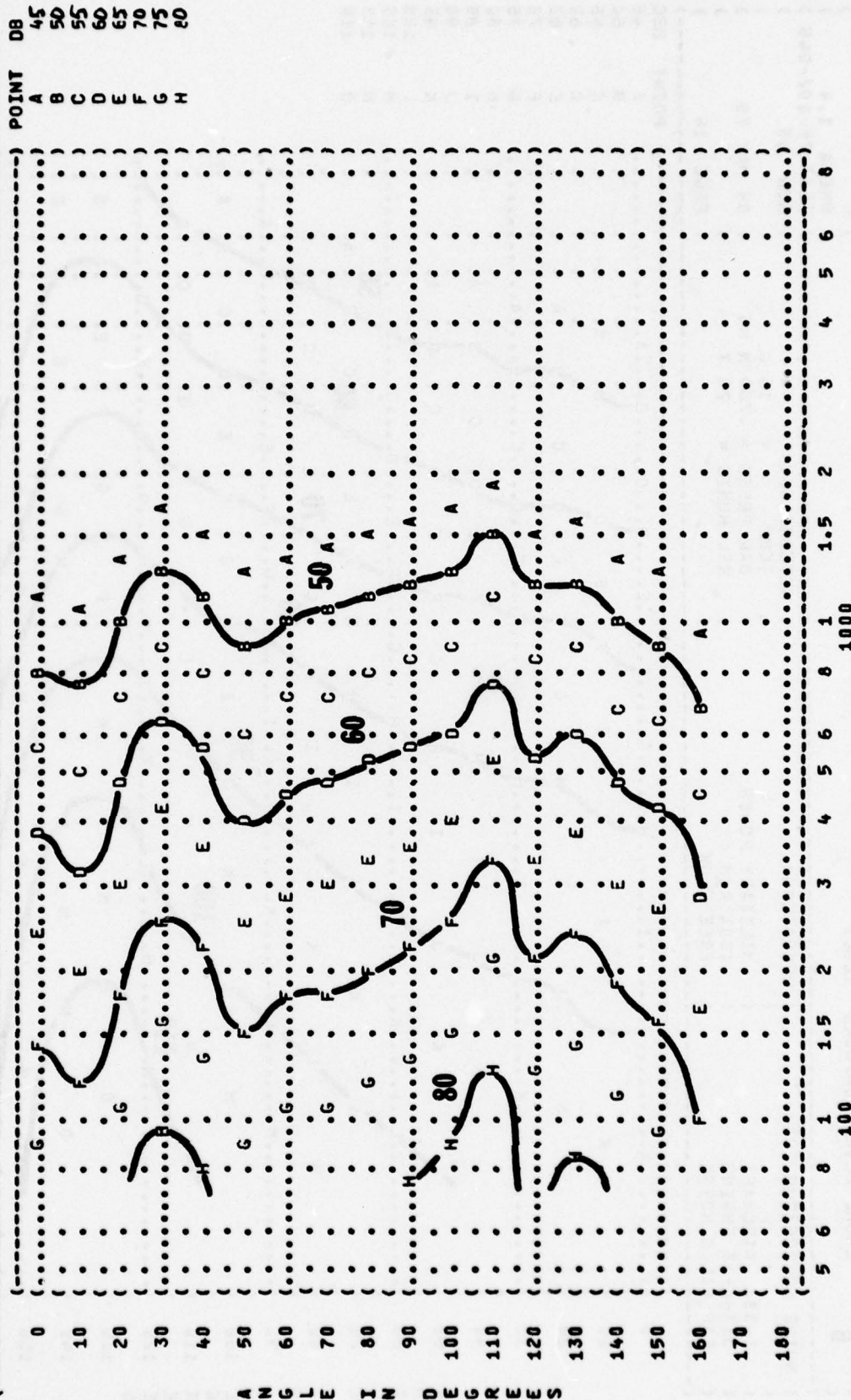
METEOROLOGY:

T-33A AIRCRAFT
J33-A-35 ENGINE
FAR FIELD NOISE

IDLE POWER
35% RPM
FREE FLOW

TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %

PAGE 15



((FIGURE: A-WEIGHTED OVERALL SOUND LEVEL (OASLA)
 ((7
 ((EQUAL LEVEL CONTOURS (DBA)
 (() IDENTIFICATION:)
 (() OMEGA 1.4
 (() TEST 75-002-045
 (() RUN 02
 (() METEOROLOGY:)
 (() TEMP = 15 C
 (() BAR PRESS = .760 M HG
 (() REL HUMID = 70 %
 (() 09 MAY 75
 (() PAGE 15
 (())

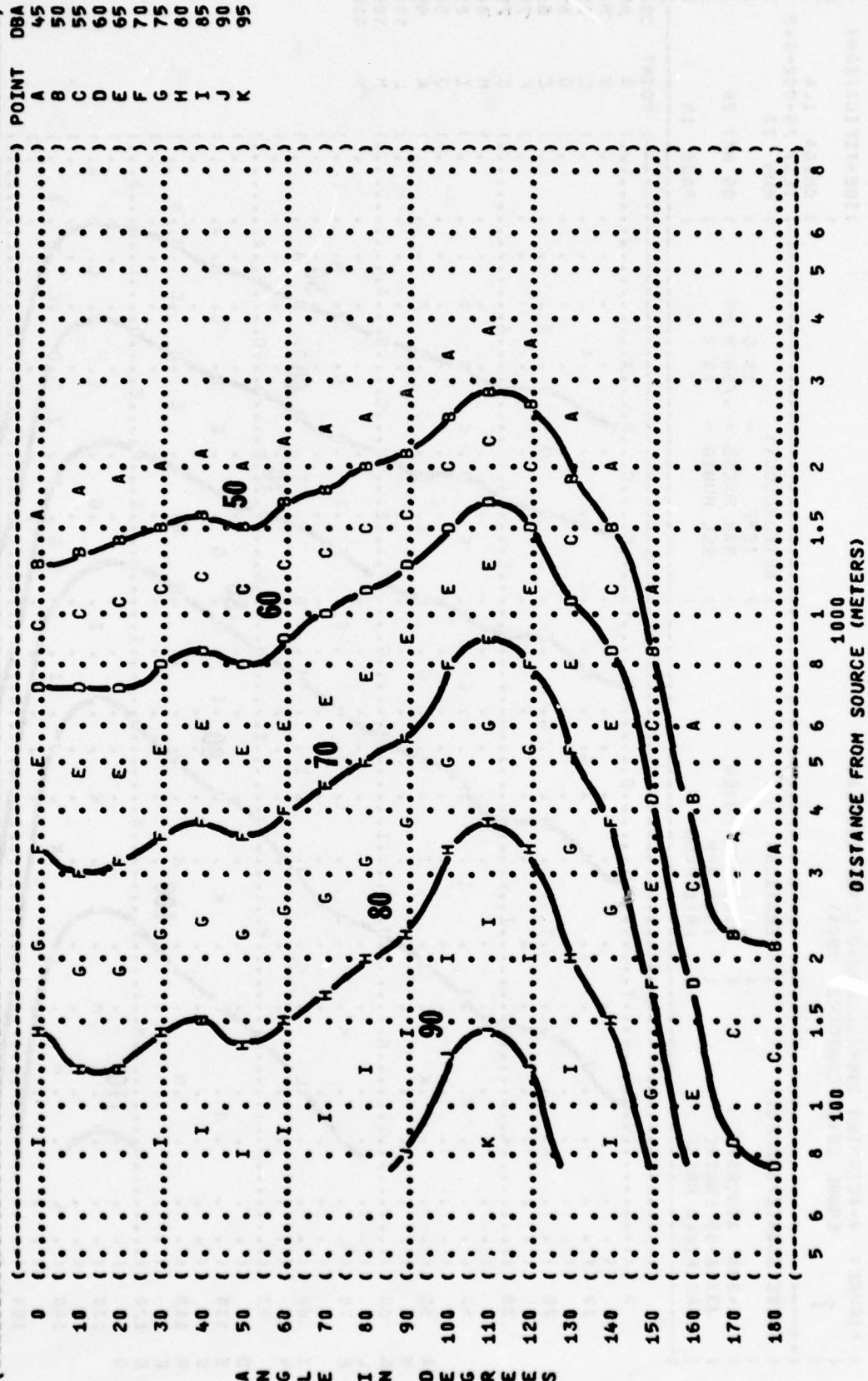


FIGURE: A-WEIGHTED OVERALL SOUND LEVEL (OASLA)
EQUAL LEVEL CONTOURS (DBA)

7

IDENTIFICATION:)
OMEGA 1.4)
TEST 75-002-045)
RUN 03)
METEOROLOGY:)
TEMP = 15 C)
BAR PRESS = .760 H HG)
REL HUMID = 70 %)
OPERATION:)
MILITARY POWER)
100% RPM)
FREE FLOW)
NOISE SOURCE/SUBJECT:)
T-33A AIRCRAFT)
J33-A-35 ENGINE)
FAR FIELD NOISE)
PAGE 15)

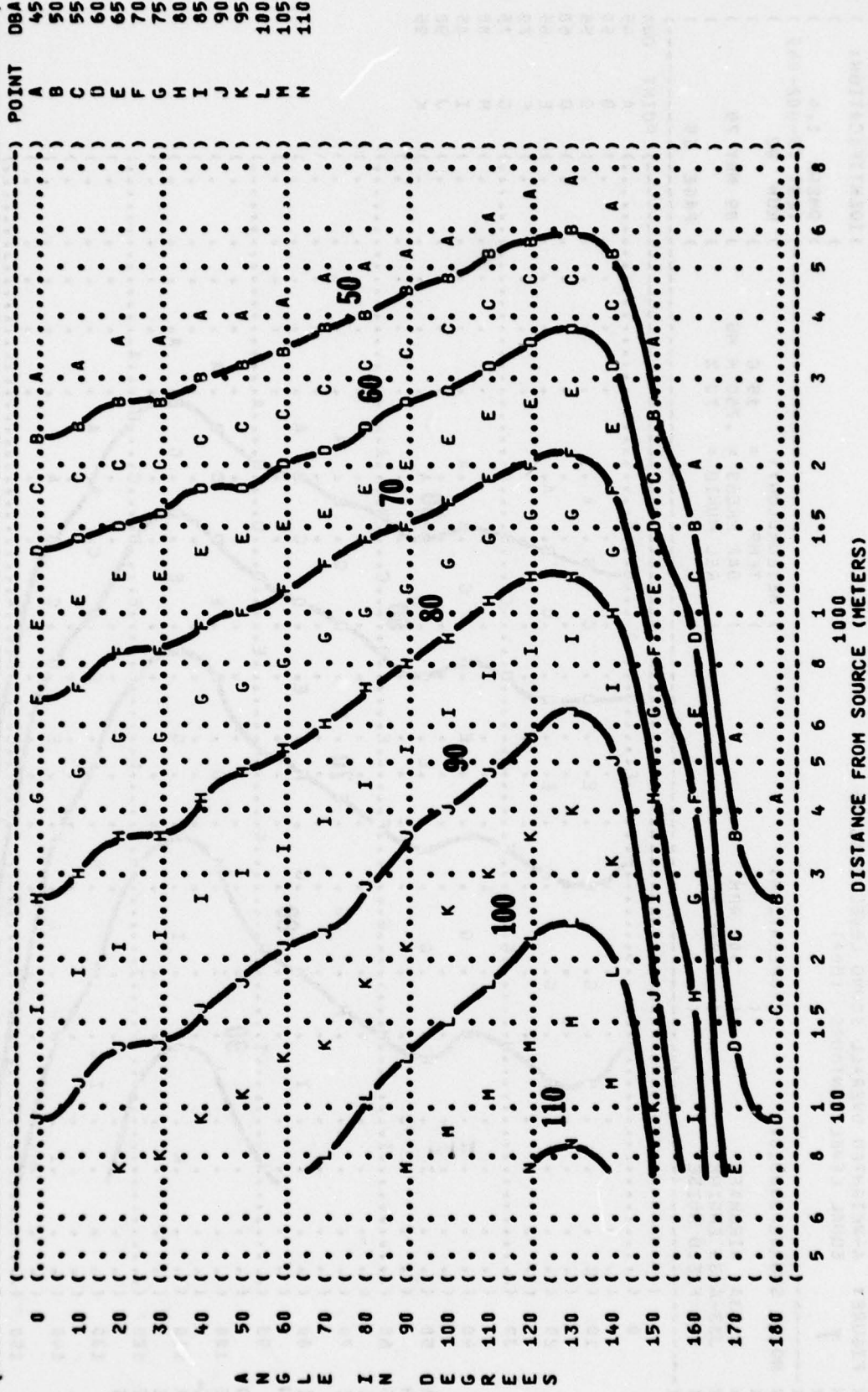
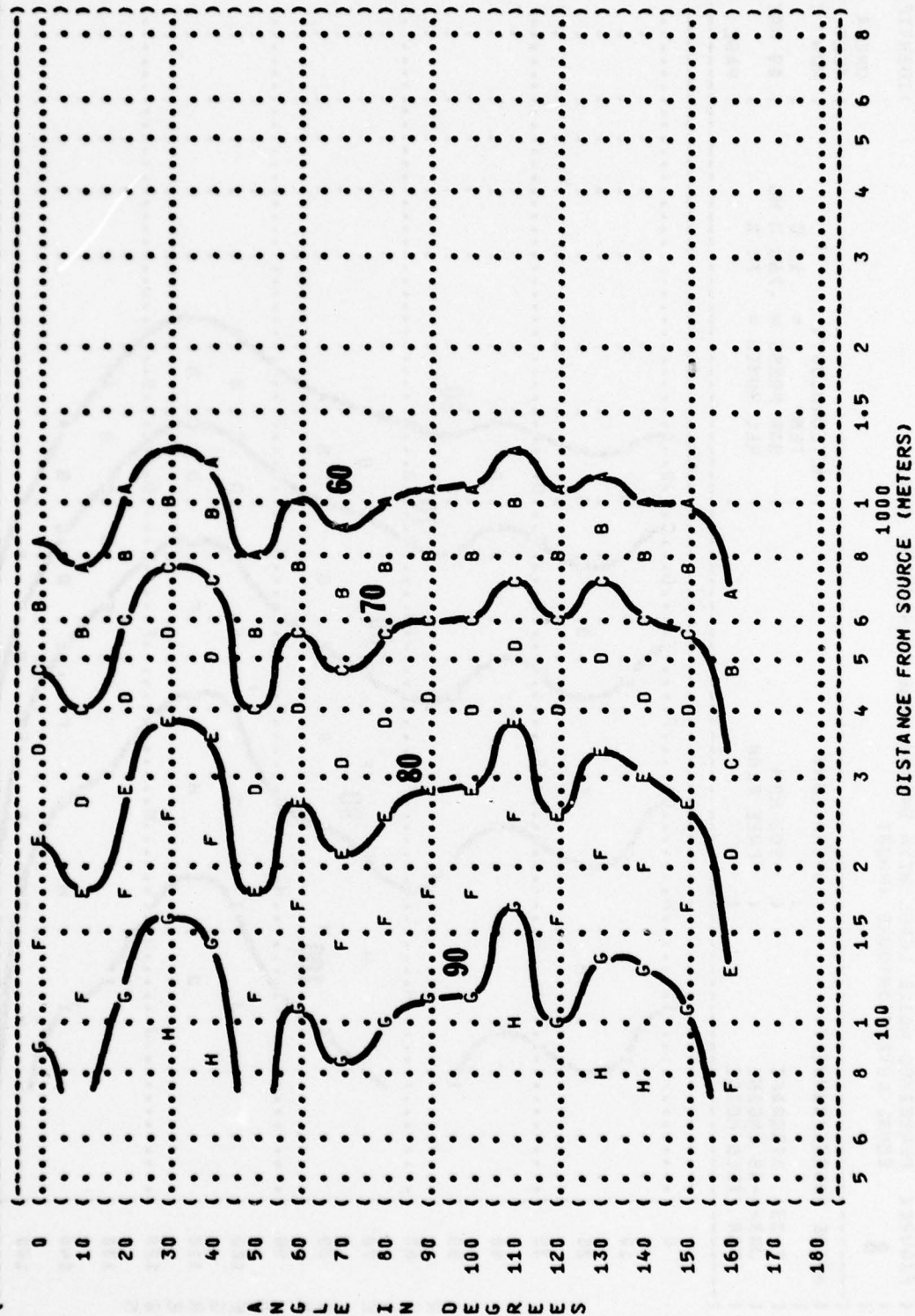


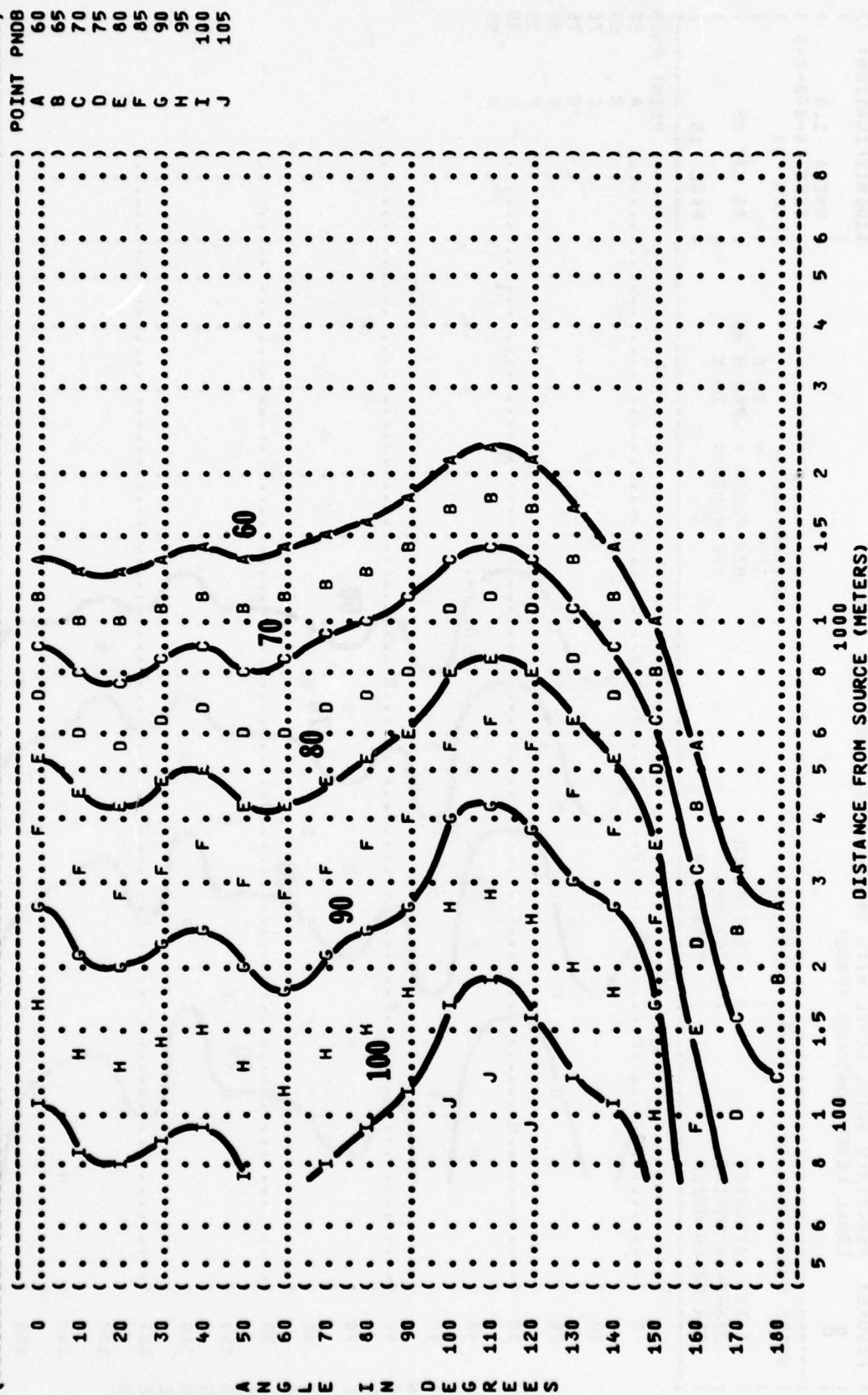
FIGURE: PERCEIVED NOISE LEVEL WITH SMOOTH TONE CORRECTION (PMLT)
 8
 EQUAL LEVEL CONTOURS (PNDB)

NOISE SOURCE/SUBJECT: () OPERATION: () METEOROLOGY: () IDENTIFICATION: ()
 () T-33A AIRCRAFT () IDLE POWER () TEMP = 15 C () OMEGA 1.4
 () J33-A-35 ENGINE () 35% RPM () BAR PRESS = .760 M HG () TEST 75-002-045
 () FAR FIELD NOISE () FREE FLOW () REL HUMID = 70 % () RUN 01
 () () () 09 MAY 75 ()
 () () () PAGE 16 ()

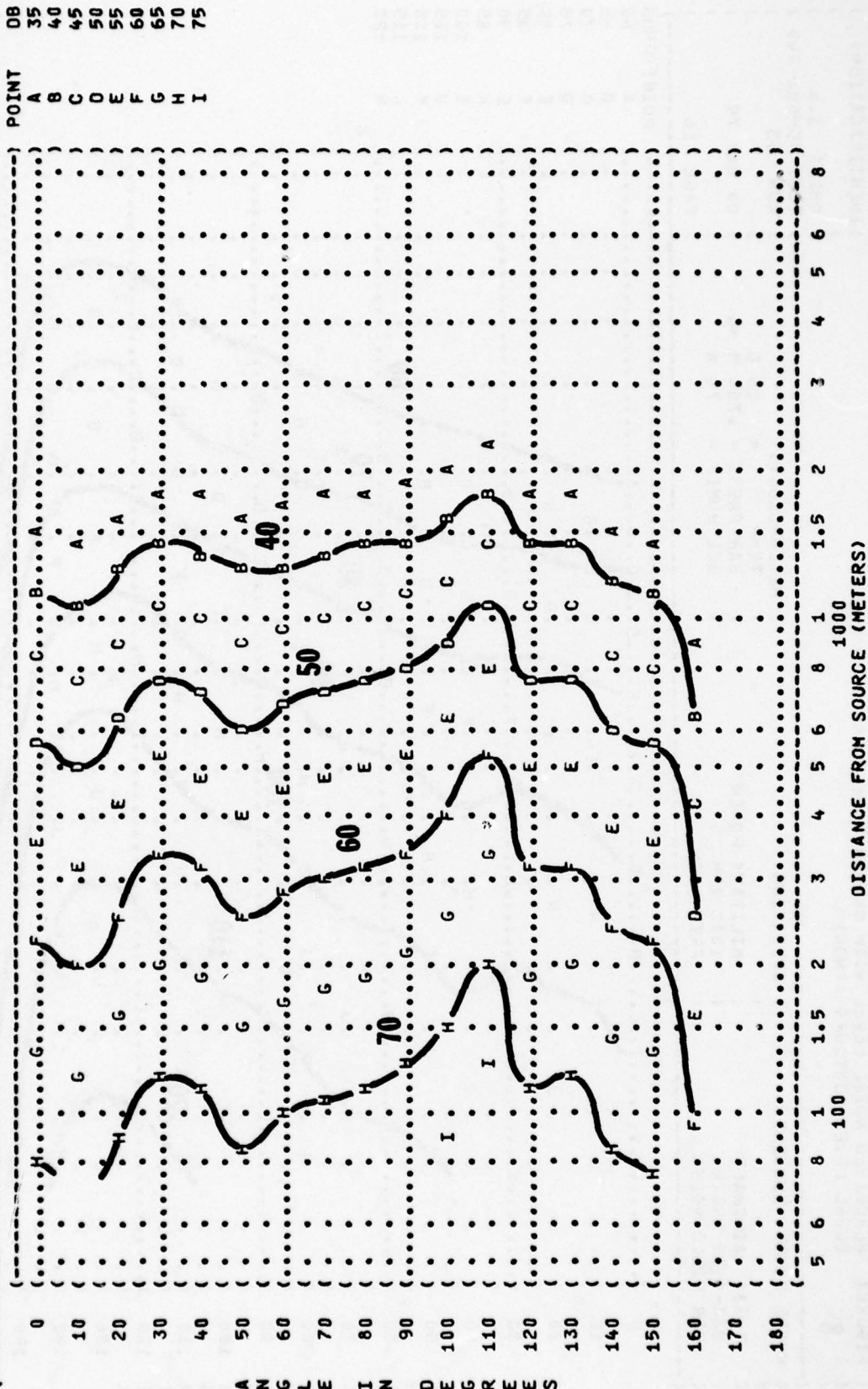
POINT PNDB
 A 60
 B 65
 C 70
 D 75
 E 80
 F 85
 G 90
 H 95




```
(-----)
( FIGURE: PERCEIVED NOISE LEVEL WITH SMOOTH TONE CORRECTION {PNLT} ) IDENTIFICATION: )
(      8    EQUAL LEVEL CONTOURS (PNOB) ) ) )
( ) ) OMEGA 1.4 )
( ) ) TEST 75-002-045 )
( ) ) RUN 02 )
( ) ) )
( ) ) TEMPERATURE = 15 C )
( ) ) BAR PRESS = .760 M HG )
( ) ) REL HUMID = 70 % )
( ) ) )
( ) ) )
( NOISE SOURCE/SUBJECT: ) OPERATION: ) METEOROLOGY: )
( ) ) )
( Y-33A AIRCRAFT ) 50% RPM ) )
( J33-A-35 ENGINE ) FREE FLOW ) )
( FAR FIELD NOISE ) ) ) PAGE 16 )
(-----)
```



() FIGURE: PREFERRED SPEECH INTERFERENCE LEVEL (PSIL)
 () 9
 () IDENTIFICATION:
 ()
 () OMEGA 1.4
 () TEST 75-002-045
 () RUN 01
 () NOISE SOURCE/SUBJECT: () OPERATION: () METEOROLOGY:
 () T-33A AIRCRAFT () TEMP = 15 C
 () J33-A-35 ENGINE () IDLE POWER () BAR PRESS = .760 M HG
 () FAR FIELD NOISE () 35% RPM () REL HUMID = 70 %
 () FREE FLOW () PAGE 17



IDENTIFICATION:

OMEGA 1.4

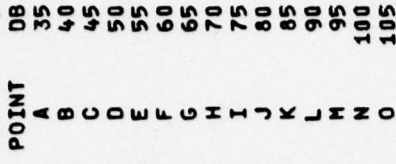
(OPERATION:

BAR PRESS = .760 H HG

REL HUMID = 70 %

PAGE 17

POINT	DB
A	35
B	40
C	45
D	50
E	55
F	60
G	65
H	70
I	75
J	80
K	85
L	90
M	95
N	100
O	105



ANGLE IN DEGREES

FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)

IDENTIFICATION:

10

OMEGA 1.4

TEST 75-002-045

RUN 01

09 MAY 75

PAGE 8

NOISE SOURCE/SUBJECT: (OPERATION:) METEOROLOGY:)

(((TEMP = 15 C)

(((IDLE POWER) BAR PRESS = .760 M HG)

(((35% RPM) REL HUMID = 70 %)

(((FREE FLOW))

PERSONNEL MAY BE EXPOSED UP TO 960 MINUTES PER DAY

AT ALL DISTANCES FROM SOURCE EQUAL TO OR GREATER THAN 75 METERS

FOR ALL ANGLES EVALUATED (INDICATED BY < AT LEFT)

UNDER THE FOLLOWING EAR PROTECTION CONDITIONS:

MINIMUM QPL EAR MUFFS

AMERICAN OPTICAL 1700 EAR MUFFS

V-51R EAR PLUGS

COMFIT TRIPLE FLANGE EAR PLUGS

H-133 GROUND COMMUNICATION UNIT

0< (

10< (

20< (

30< (

40< (

50< (

60< (

70< (

80< (

90< (

100< (

110< (

120< (

130< (

140< (

150< (

160< (

170 (

180 (

5 6 8 1 1.5 2 3 4 5 6 8

100 1000

DISTANCE FROM SOURCE (METERS)

A N G L E I N D E R E S

FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)

IDENTIFICATION:

10 EQUAL TIME CONTOURS (MINUTES)

NO PROTECTION

NOISE SOURCE/SUBJECT: (OPERATION:) METEOROLOGY:)

T-33A AIRCRAFT (50% RPM) TEMP = 15 C)

J33-A-35 ENGINE (FREE FLOW) BAR PRESS = .760 H HG)

FAR FIELD NOISE () REL HUMID = 70 %)

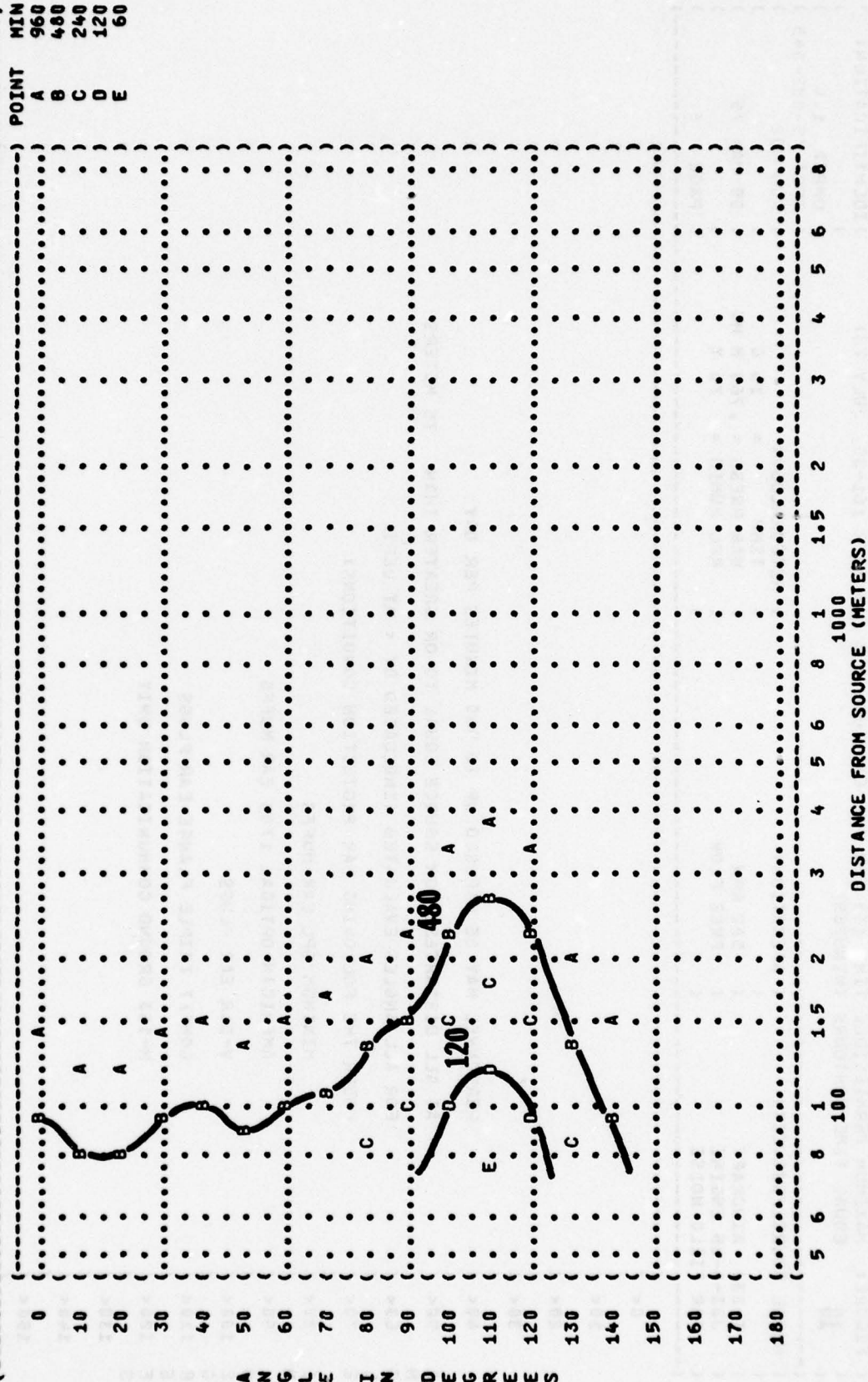
OMEGA 1.4)

TEST 75-002-045)

RUN 02)

09 MAY 75)

PAGE 7)



```
(-----)
( FIGURE: MAXIMUM PERMISSIBLE TIME {T} FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73) ) IDENTIFICATION: )
(      10      EQUAL TIME CONTOURS (MINUTES) ) )
(-----)
( NOISE SOURCE/SUBJECT: ) OPERATION: ) METEOROLOGY: ) OMEGA 1.4 )
( T-33A AIRCRAFT ) 50% RPM ) TEMP = 15 C ) TEST 75-002-045 )
( J33-A-35 ENGINE ) FREE FLOW ) BAR PRESS = .760 M HG ) RUN 02 )
( FAR FIELD NOISE ) ) REL HUMID = 70 % ) PAGE 8 )
(-----)
```

PERSONNEL MAY BE EXPOSED UP TO 960 MINUTES PER DAY
AT ALL DISTANCES FROM SOURCE EQUAL TO OR GREATER THAN 75 METERS
FOR ALL ANGLES EVALUATED (INDICATED BY < AT LEFT)
UNDER THE FOLLOWING EAR PROTECTION CONDITIONS:

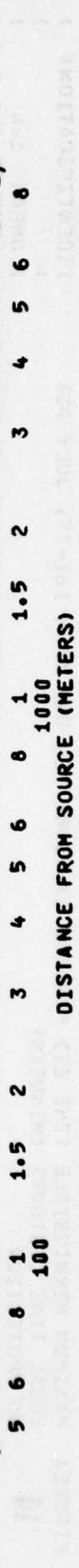
MINIMUM QPL EAR MUFFS
AMERICAN OPTICAL 1700 EAR MUFFS
V-51R EAR PLUGS
COMFIT TRIPLE FLANGE EAR PLUGS
H-133 GROUND COMMUNICATION UNIT

5 6 0 1 1.5 2 3 4 5 6 0
100 1000
DISTANCE FROM SOURCE (METERS)

	MIN	POINT
0	960	A
10	480	B
20	240	C
30	120	D
	60	E
	30	F
	15	G
	8	H

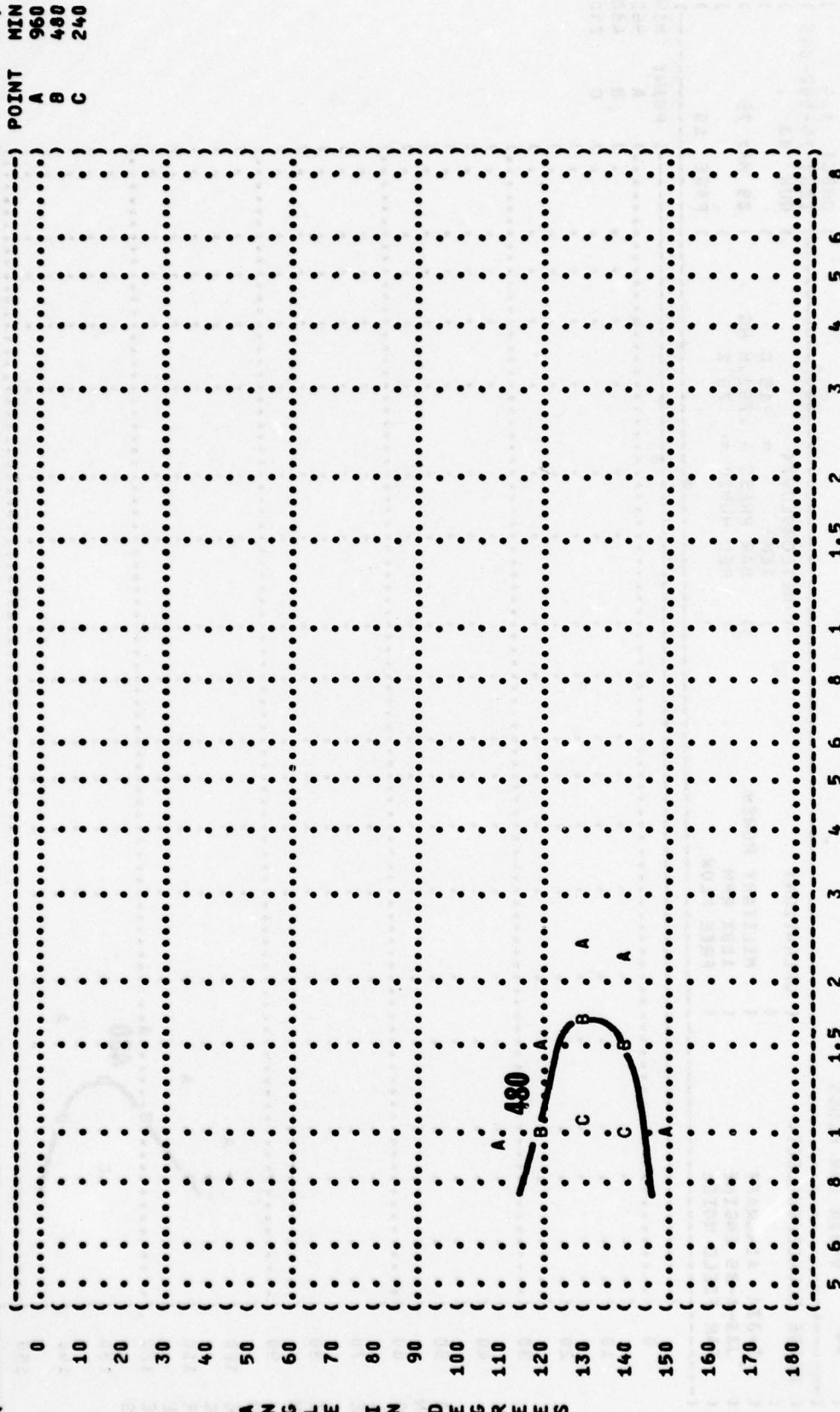
The graph displays the relationship between the Angle of Incidence (X-axis) and the Point (Y-axis). The X-axis ranges from 0 to 180 degrees, and the Y-axis ranges from A to E. Two curves are plotted: one labeled '120' and another labeled '480'. The curves show that for a given angle of incidence, there are multiple points (A, B, C, D, E) corresponding to different values of the Y-axis variable. The curves are symmetric about 90 degrees.

Angle of Incidence (Degrees)	Point (120)	Point (480)
0	A	A
10	B	B
20	C	C
30	D	D
40	E	E
50	A	A
60	B	B
70	C	C
80	D	D
90	E	E
100	A	A
110	B	B
120	C	C
130	D	D
140	E	E
150	A	A
160	B	B
170	C	C
180	D	D



1000

(FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)) IDENTIFICATION:)
 (10) EQUAL TIME CONTOURS (MINUTES))
 (AMERICAN OPTICAL 1700 EAR MUFFS) OMEGA 1.4
 () TEST 75-002-045)
 (NOISE SOURCE/SUBJECT:) OPERATION:) METEOROLOGY:) RUN 03)
 ()))
 (T-33A AIRCRAFT)) TEMP = 15 C))
 (J33-A-35 ENGINE)) BAR PRESS = .760 M HG) 09 MAY 75)
 (FAR FIELD NOISE)) REL HUMID = 70 %))
 ()))) PAGE 9)



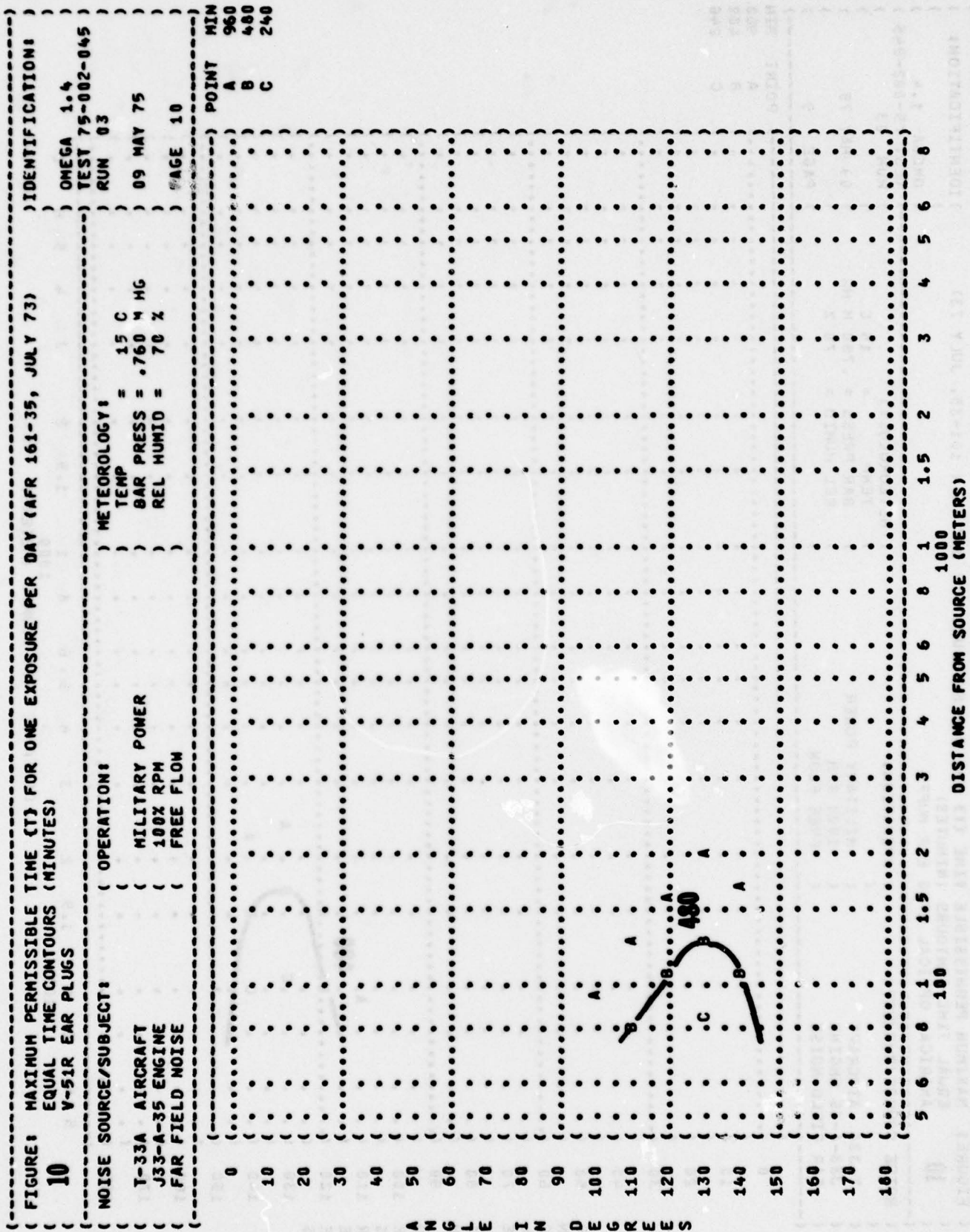


FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)
EQUAL TIME CONTOURS (MINUTES)
CONFIT TRIPLE FLANGE EAR PLUGS

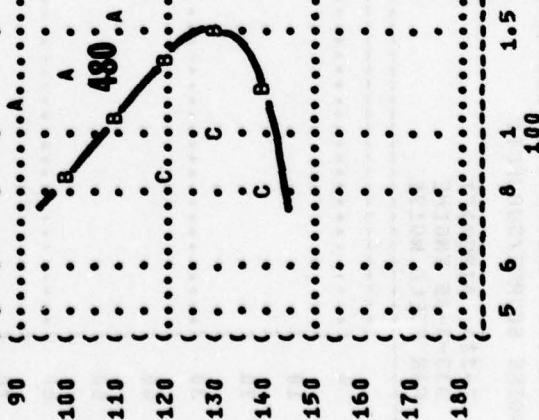
10

) IDENTIFICATIONS
)
) OMEGA 1.4

NOISE SOURCE/SUBJECT:	(OPERATION:	METEOROLOGY:	TEMP =	15 C	09 MAY 75
T-33A AIRCRAFT	(MILITARY POWER	(BAR PRESS =	.760 M HG		
J33-A-35 ENGINE	(100% RPM	(REL HUMID =	70 %		
FAR FIELD NOISE	(FREE FLOW	(PAGE 11

	POINT
0	A
10	B
	C

ANGLE IN DEGREES

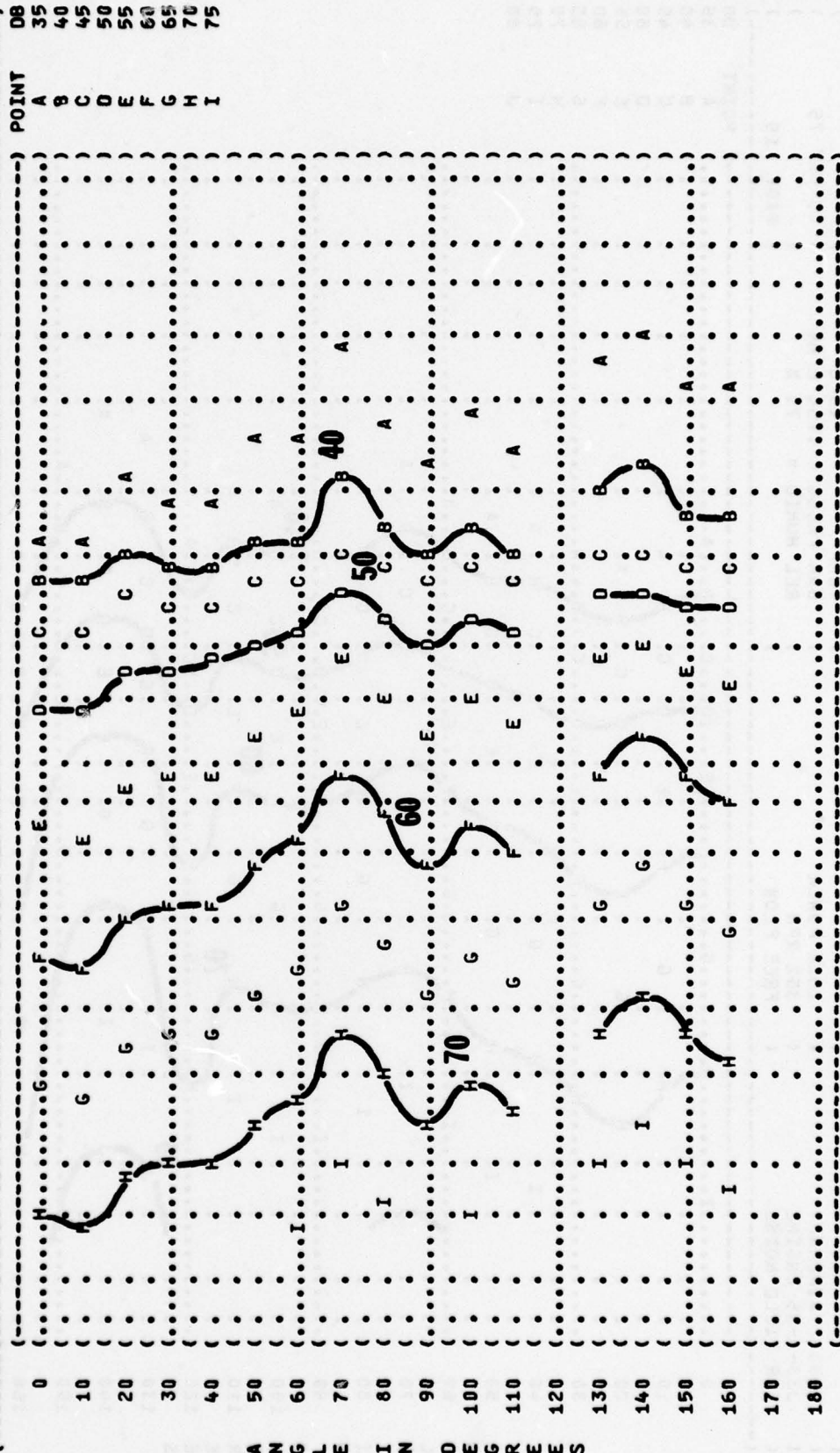


DISTANCE FROM SOURCE (METERS)

ANGIE IN DEGREWS



(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (EQUAL LEVEL CONTOURS (DB)
 (11 31.5 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION:
 (T-33A AIRCRAFT (IDLE POWER
 (J33-A-35 ENGINE (35% RPM
 (FAR FIELD NOISE (FREE FLOW
 (METEOROLOGY: (TEMP = 15 C
 (BAR PRESS = .760 M HG
 (REL HUMID = 70 %
 (RUN 01
 (09 MAY 75
 (PAGE 18
 (IDENTIFICATION:
 (OMEGA 1.4
 (TEST 75-002-045
 (



A N G L E I N D E G R E E S

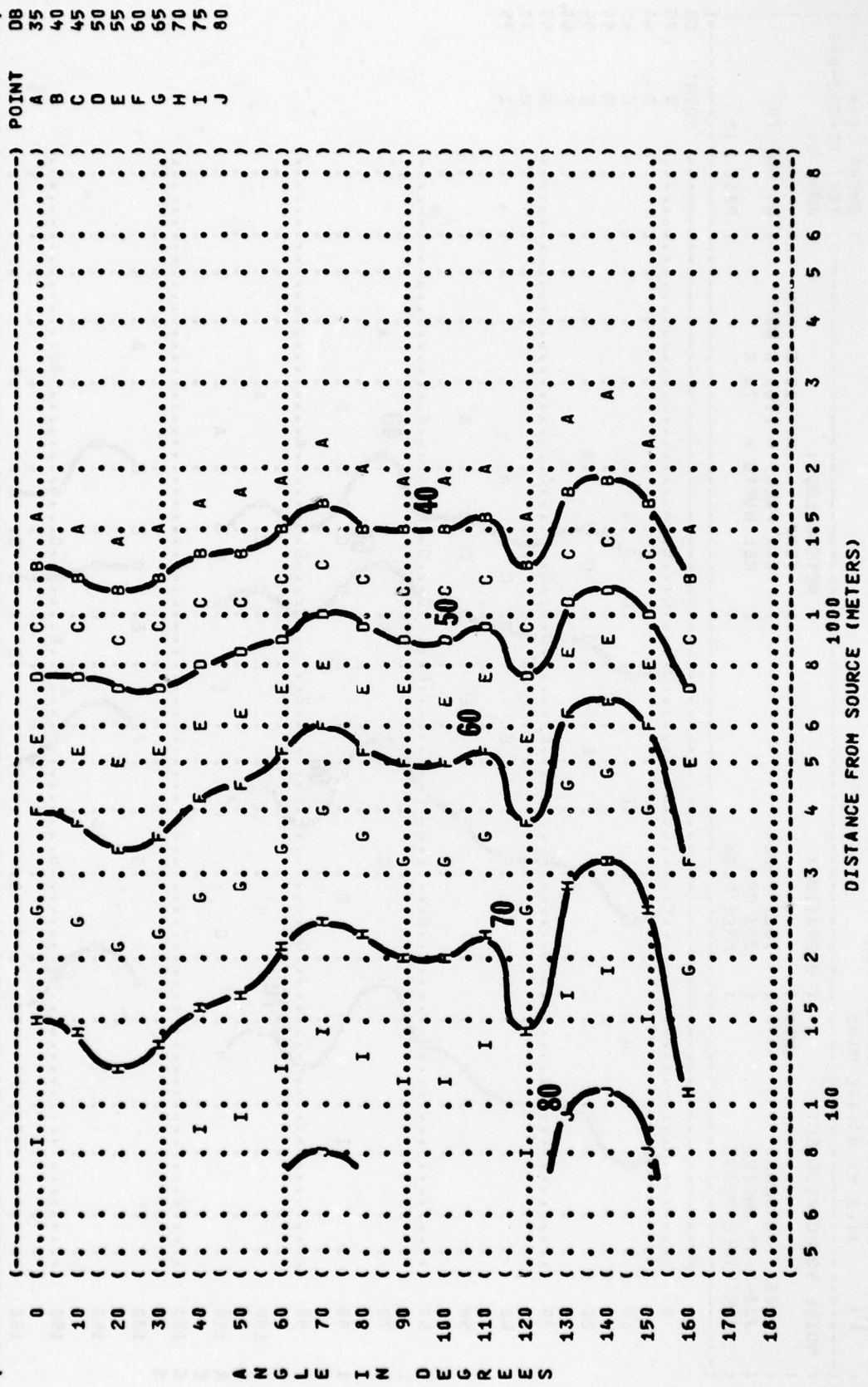
FIGURE 11 SOUND PRESSURE LEVEL (SPL) EQUAL LEVEL CONTOURS (DB) 63 HZ OCTAVE BAND

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-045
RUN 01

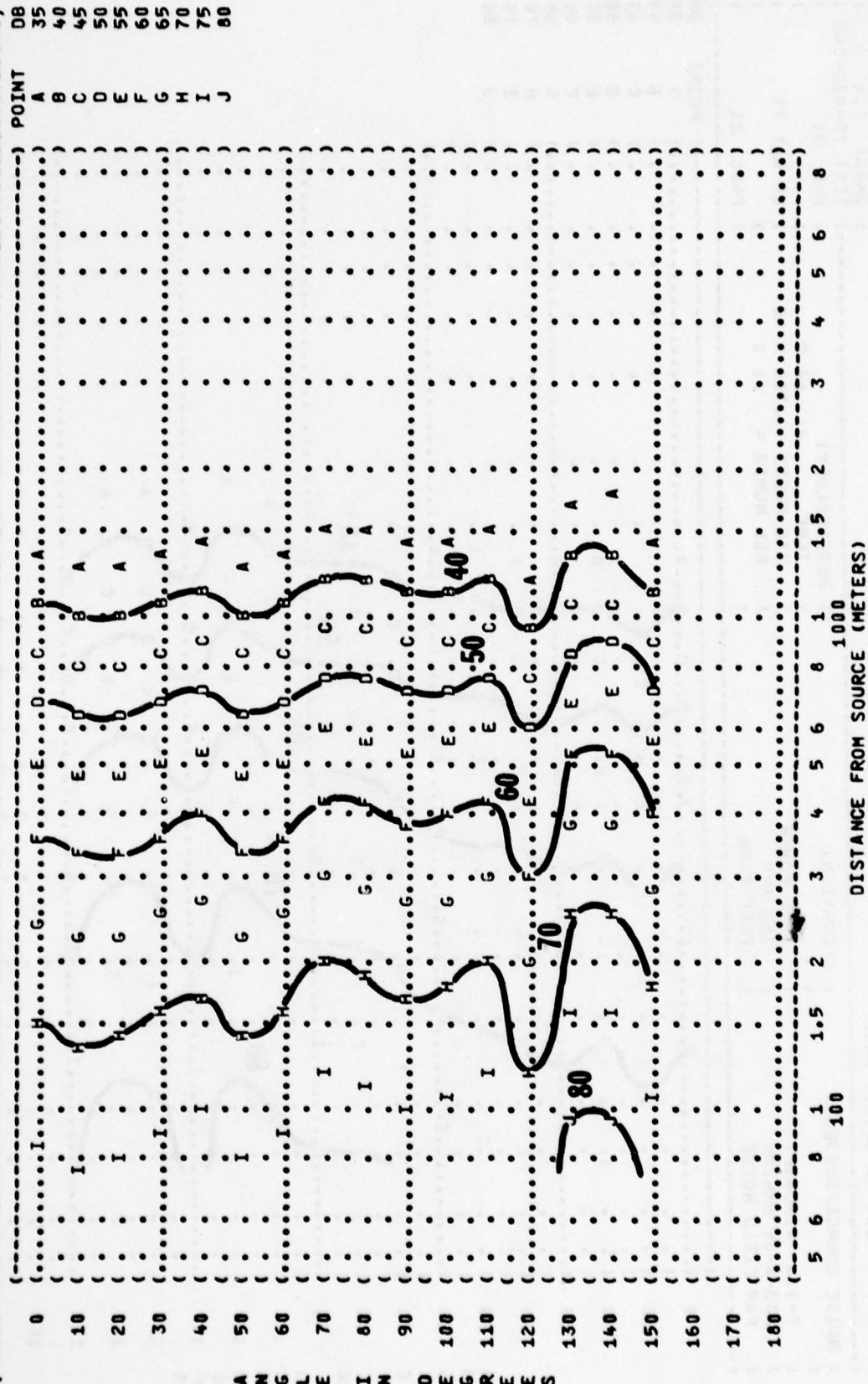
NOISE SOURCE/SUBJECT:	OPERATION:
T-33A AIRCRAFT	IDLE POWER
J33-A-35 ENGINE	35% RPM
FAR FIELD NOISE	FREE FLOW

) METEOROLOGY:
) TEMP = 15 C
) BAR PRESS = .760 M HG
) REL HUMID = 70 %
)

PAGE 19



(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (11 EQUAL LEVEL CONTOURS (DB)
 (125 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT:
 ((OPERATION:
 (((IDLE POWER
 (((35% RPM
 (((FREE FLOW
 (((METEOROLOGY:
 (((TEMP = 15 C
 (((BAR PRESS = .760 M HG
 (((REL HUMID = 70 %
 (((09 MAY 75
 (((PAGE 20
 (IDENTIFICATION:
 (OMEGA 1.4
 (TEST 75-002-045
 (RUN 01



A N G L E I M D E G R E E S

FIGURE 1 SOUND PRESSURE LEVEL (SPL)
EQUAL LEVEL CONTOURS (DB)
250 HZ OCTAVE BAND

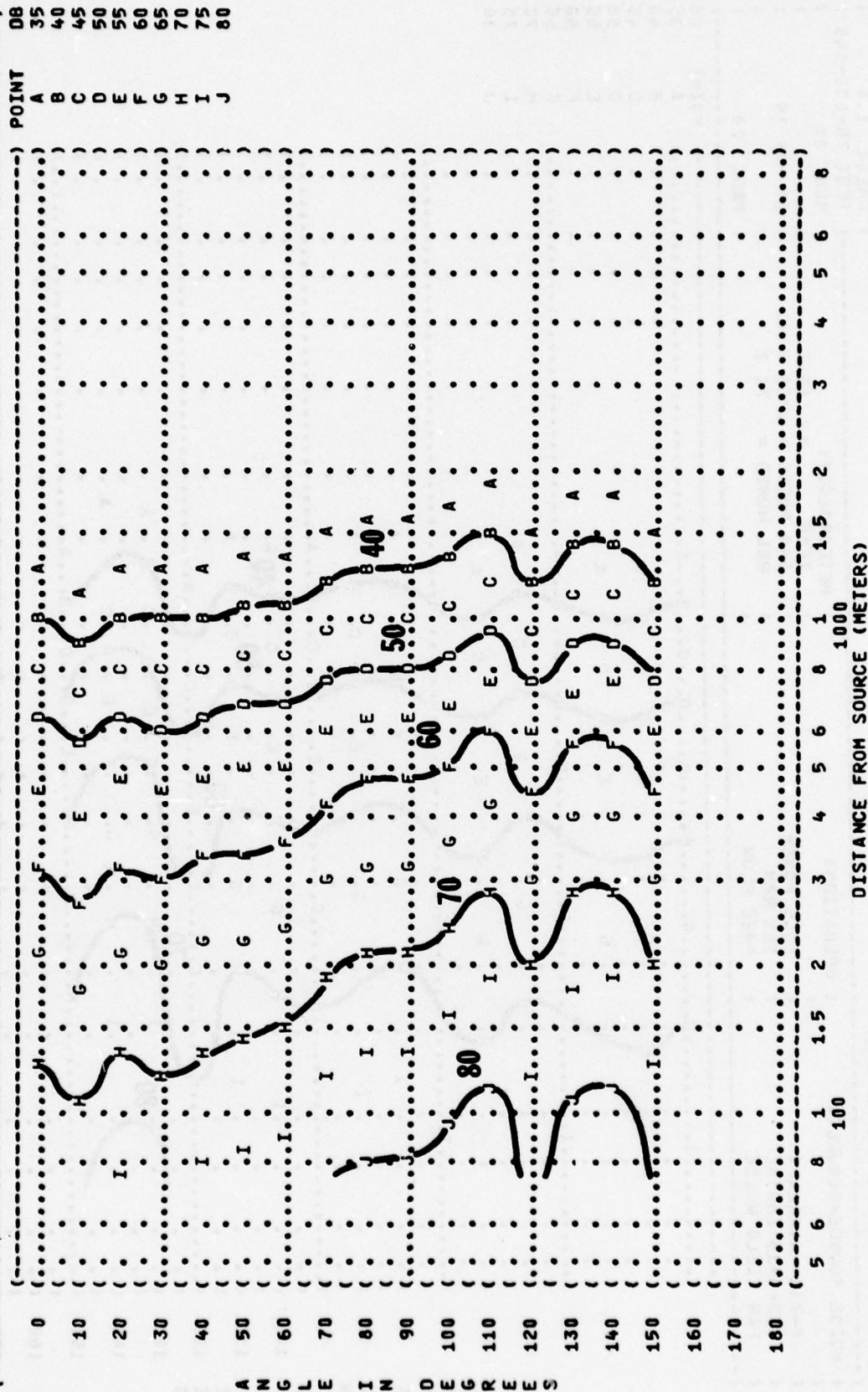
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IDENTIFICATION:
OMEGA 1.4
TEST 75-002-045
RUN 01

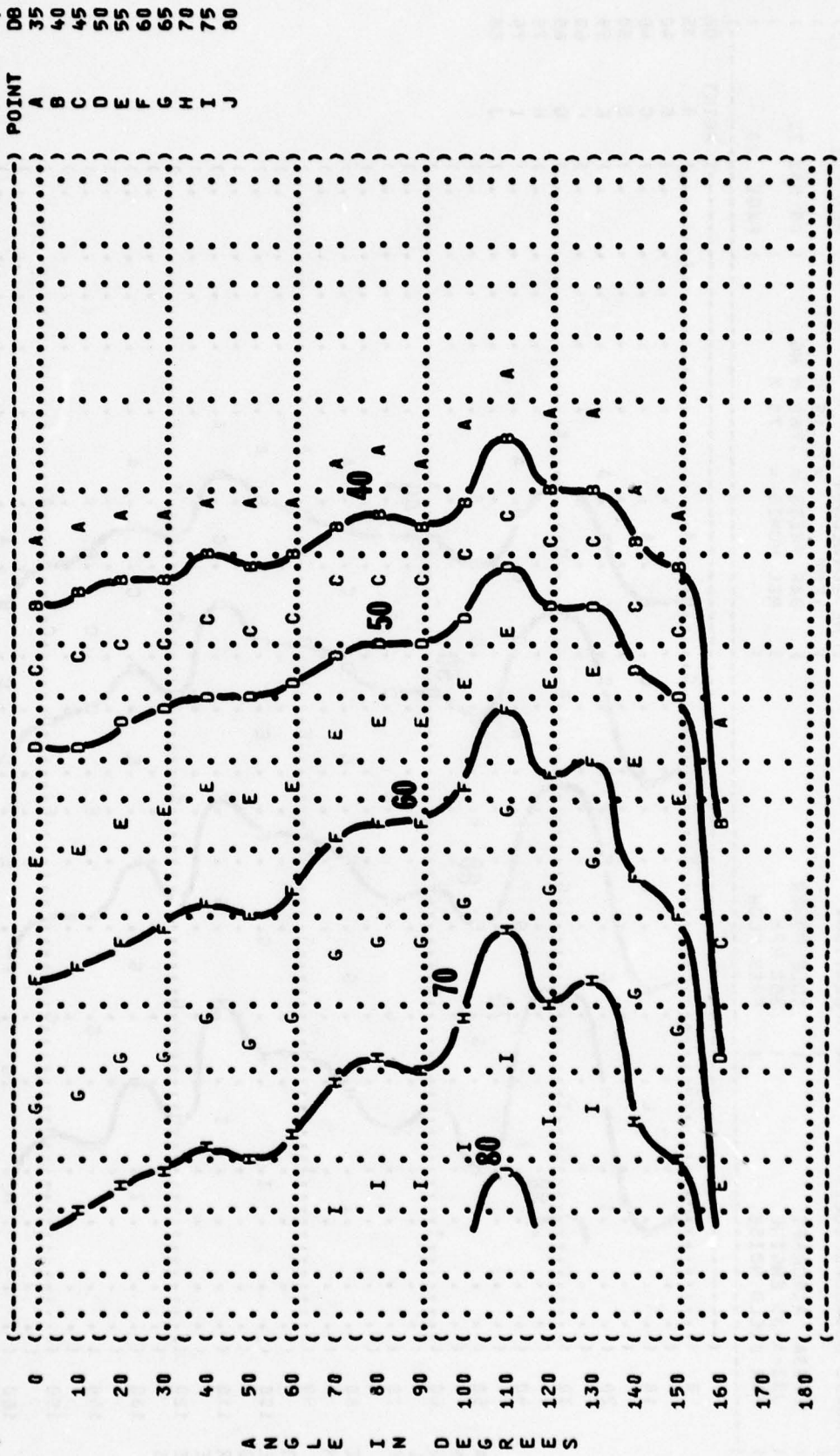
NOISE SOURCE/SUBJECT: OPERATION:
T-33A AIRCRAFT
J33-A-35 ENGINE
FAR FIELD NOISE

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %

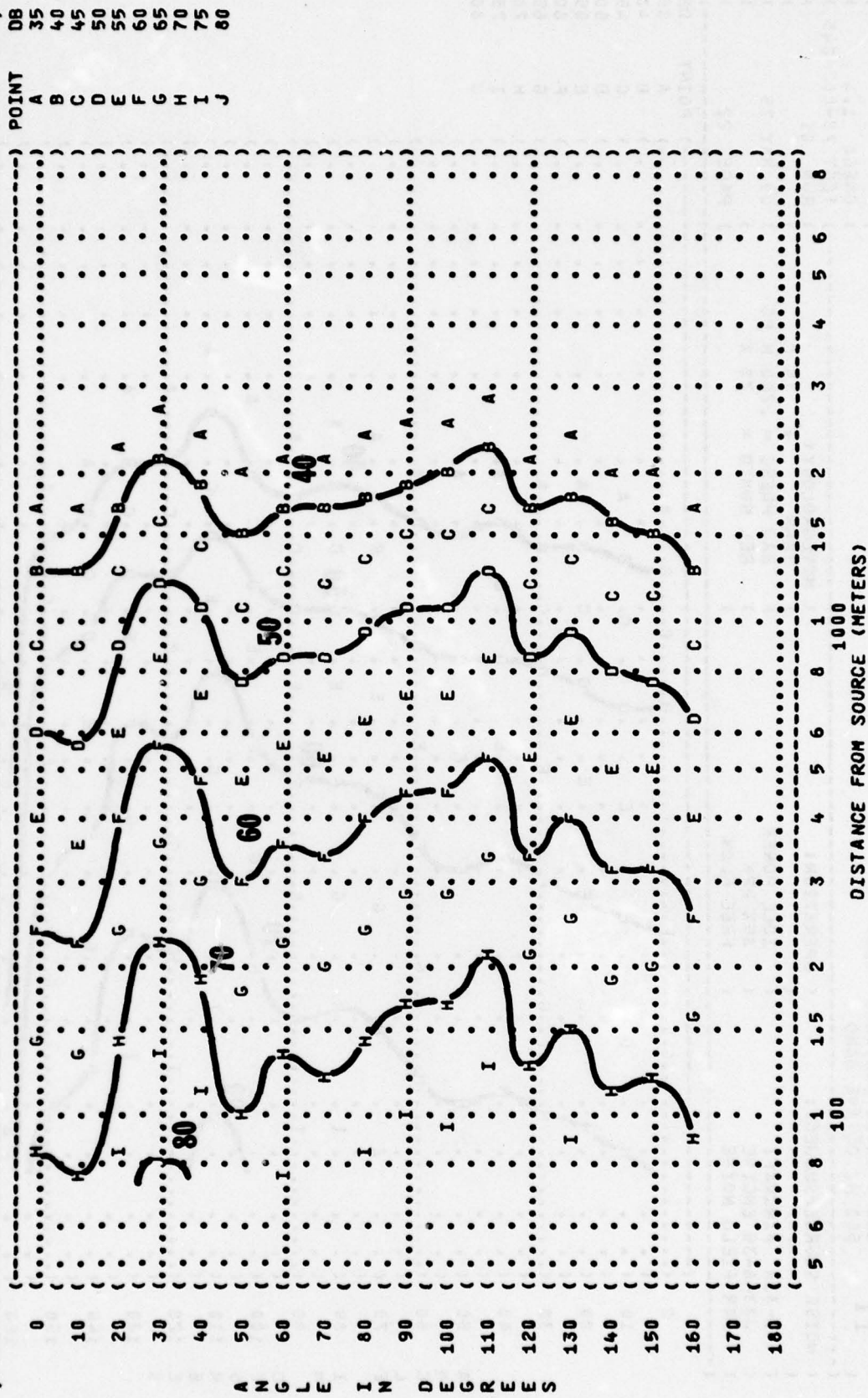
PAGE 21



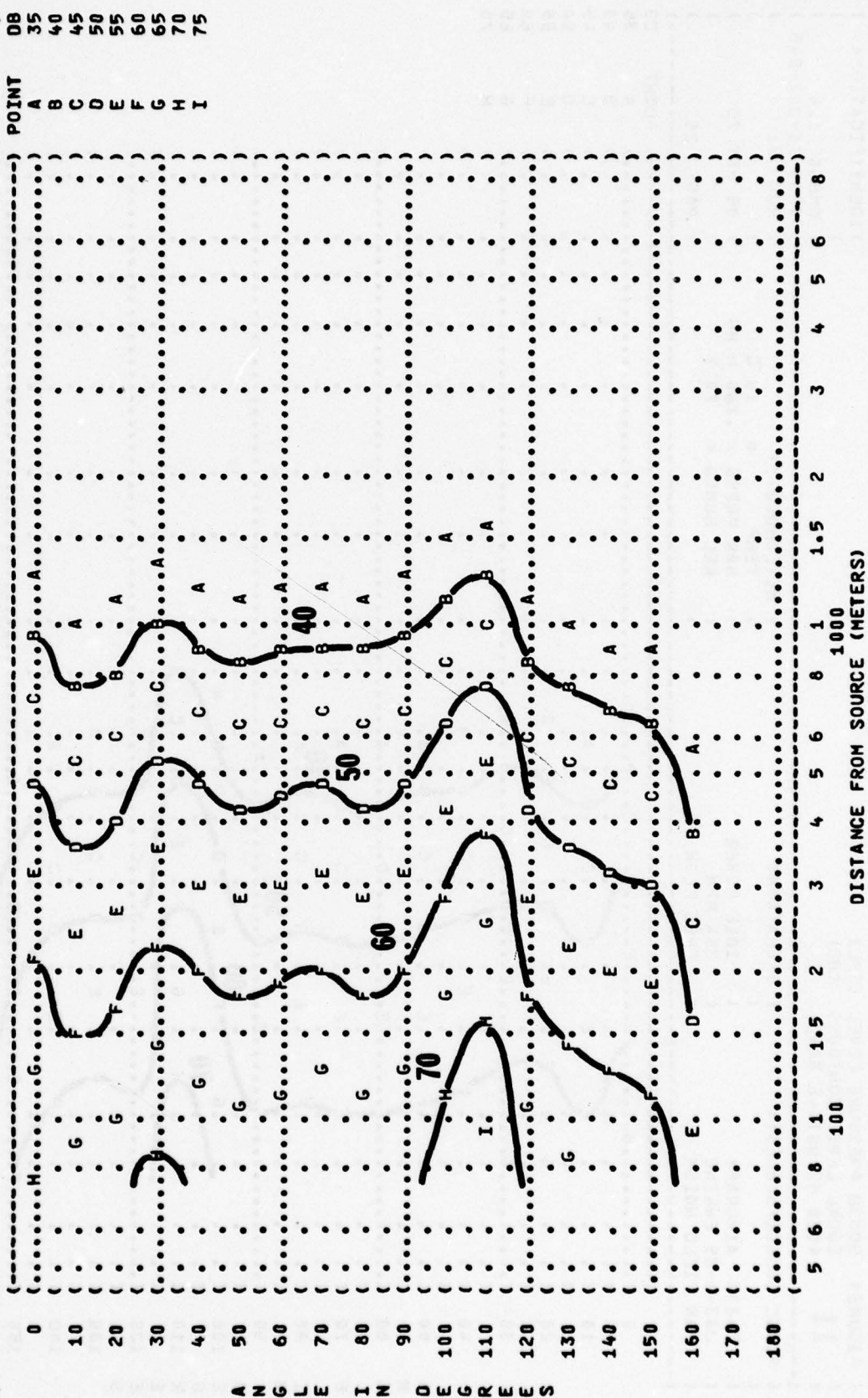
(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (11 EQUAL LEVEL CONTOURS (DB)
 (500 HZ OCTAVE BAND
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 (T-33A AIRCRAFT (IDLE POWER
 (J33-A-35 ENGINE (35% RPM
 (FAR FIELD NOISE (FREE FLOW
 (METEOROLOGY:
 (TEMP = 15 C
 (BAR PRESS = .760 M HG
 (REL HUMID = 70 %
 (IDENTIFICATION:
 (OMEGA 1.4
 (TEST 75-002-045
 (RUN 01
 (09 MAY 75
 (PAGE 22



(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (11 EQUAL LEVEL CONTOURS (DB)
 (1000 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT:
 (T-33A AIRCRAFT
 (J33-A-35 ENGINE
 (FAR FIELD NOISE
 (OPERATION:
 (IDLE POWER
 (35% RPM
 (FREE FLOW
 (METEOROLOGY:
 (TEMP = 15 C
 (BAR PRESS = .760 M HG
 (REL HUMID = 70 %
 (IDENTIFICATION:
 (OMEGA 1.4
 (TEST 75-002-045
 (RUN 01
 (09 MAY 75
 (PAGE 23

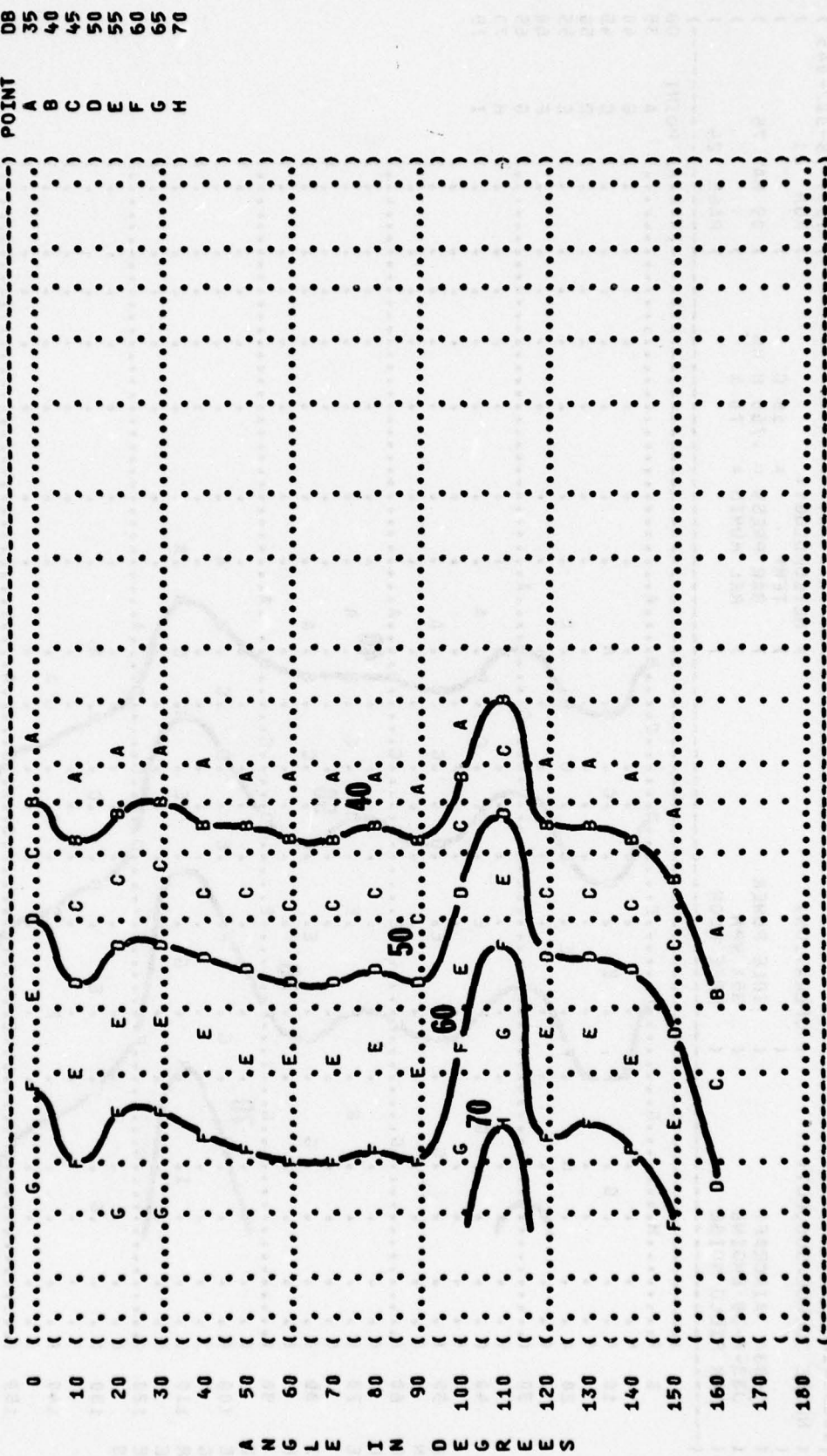


(FIGURE: SOUND PRESSURE LEVEL (SPL))
 (11 EQUAL LEVEL CONTOURS (DB))
 (2000 HZ OCTAVE BAND)
 (NOISE SOURCE/SUBJECT:)
 (T-33A AIRCRAFT)
 (J33-A-35 ENGINE)
 (FAR FIELD NOISE)
 (OPERATION:)
 (IDLE POWER)
 (35% RPM)
 (FREE FLOW)
 (METEOROLOGY:)
 (TEMP = 15 C)
 (BAR PRESS = .760 H HG)
 (REL HUMID = 70 %)
 (IDENTIFICATION:)
 (OMEGA 1.4)
 (TEST 75-002-045)
 (RUN 01)
 (09 MAY 75)
 (PAGE 24)



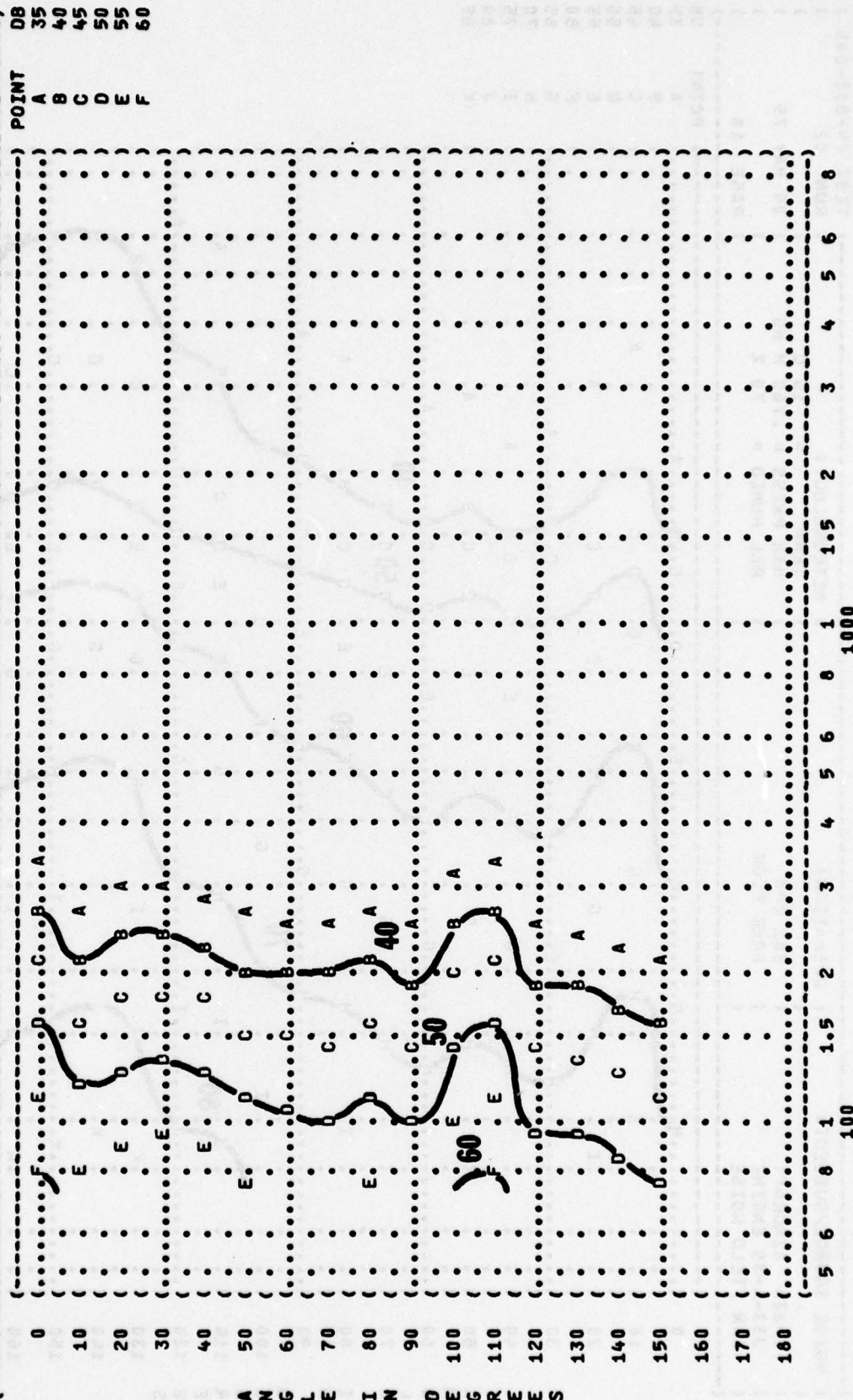
A N G L E I N D E G R E E S

(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (11 EQUAL LEVEL CONTOURS (DB)
 (4000 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION:
 (T-33A AIRCRAFT (IDLE POWER
 (J33-A-35 ENGINE (35% RPM
 (FAR FIELD NOISE (FREE FLOW
 (METEOROLOGY: TEMP = 15 C
 (BAR PRESS = .760 H HG
 (REL HUMID = 70 %
 (IDENTIFICATION: OMEGA 1.4
 (TEST 75-002-045
 (RUN 01
 (09 MAY 75
 (PAGE 25



DB 35
 40
 45
 50
 55
 60
 65
 70
 POINT A B C D E F G H
 DISTANCE FROM SOURCE (METERS)

(FIGURE: SOUND PRESSURE LEVEL (SPL)) IDENTIFICATION:)
 (11 EQUAL LEVEL CONTOURS (DB)))
 (8000 HZ OCTAVE BAND))
 (NOISE SOURCE/SUBJECT:))
 (T-33A AIRCRAFT))
 (J33-A-35 ENGINE))
 (FAR FIELD NOISE))
 (OPERATION:))
 (IDLE POWER))
 (35% RPM))
 (FREE FLOW))
 (METEOROLOGY:))
 (TEMP = 15 C))
 (BAR PRESS = .760 M HG))
 (REL HUMID = 70 %))
 (RUN 01))
 (TEST 75-002-045))
 (PAGE 26))



A N G L E I N D E G R E E S

FIGURE: SOUND PRESSURE LEVEL (SPL)
EQUAL LEVEL CONTOURS (DB)
31.5 HZ OCTAVE BAND

11

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-045
RUN 02

NOISE SOURCE/SUBJECT:

OPERATION:

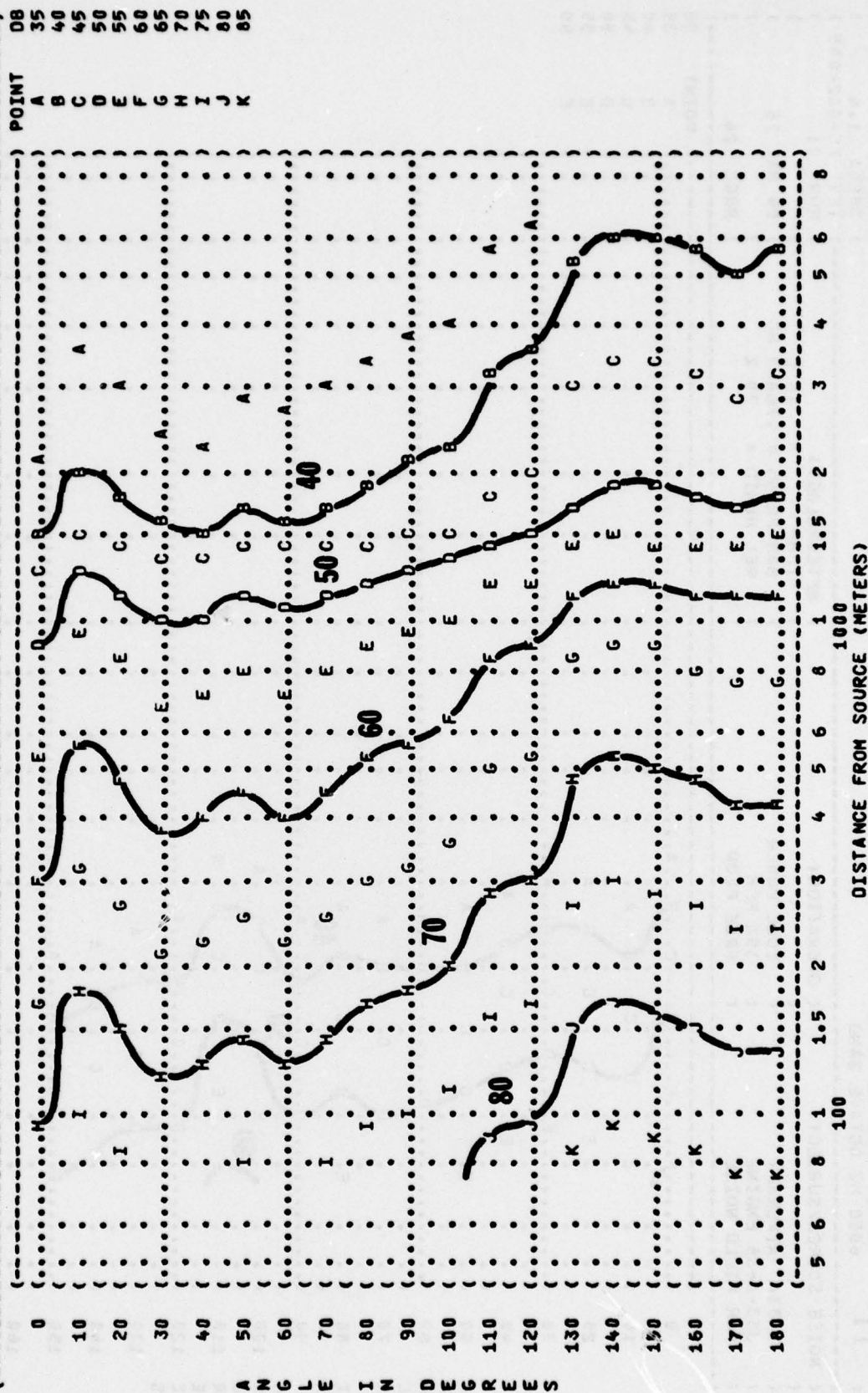
METEOLOGY:

T-33A AIRCRAFT
J33-A-35 ENGINE
FAR FIELD NOISE

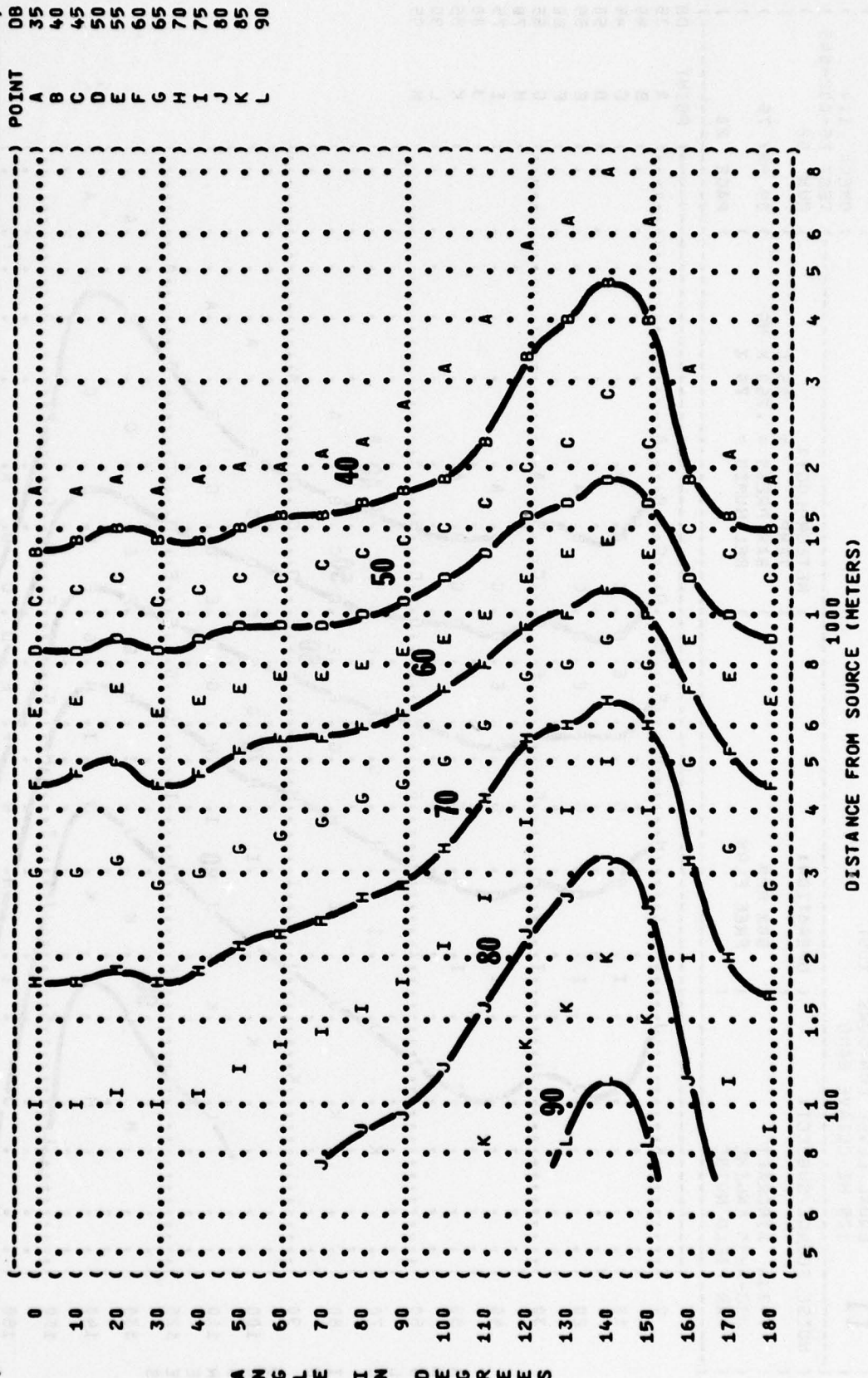
50% RPM
FREE FLOW

TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %

PAGE 18

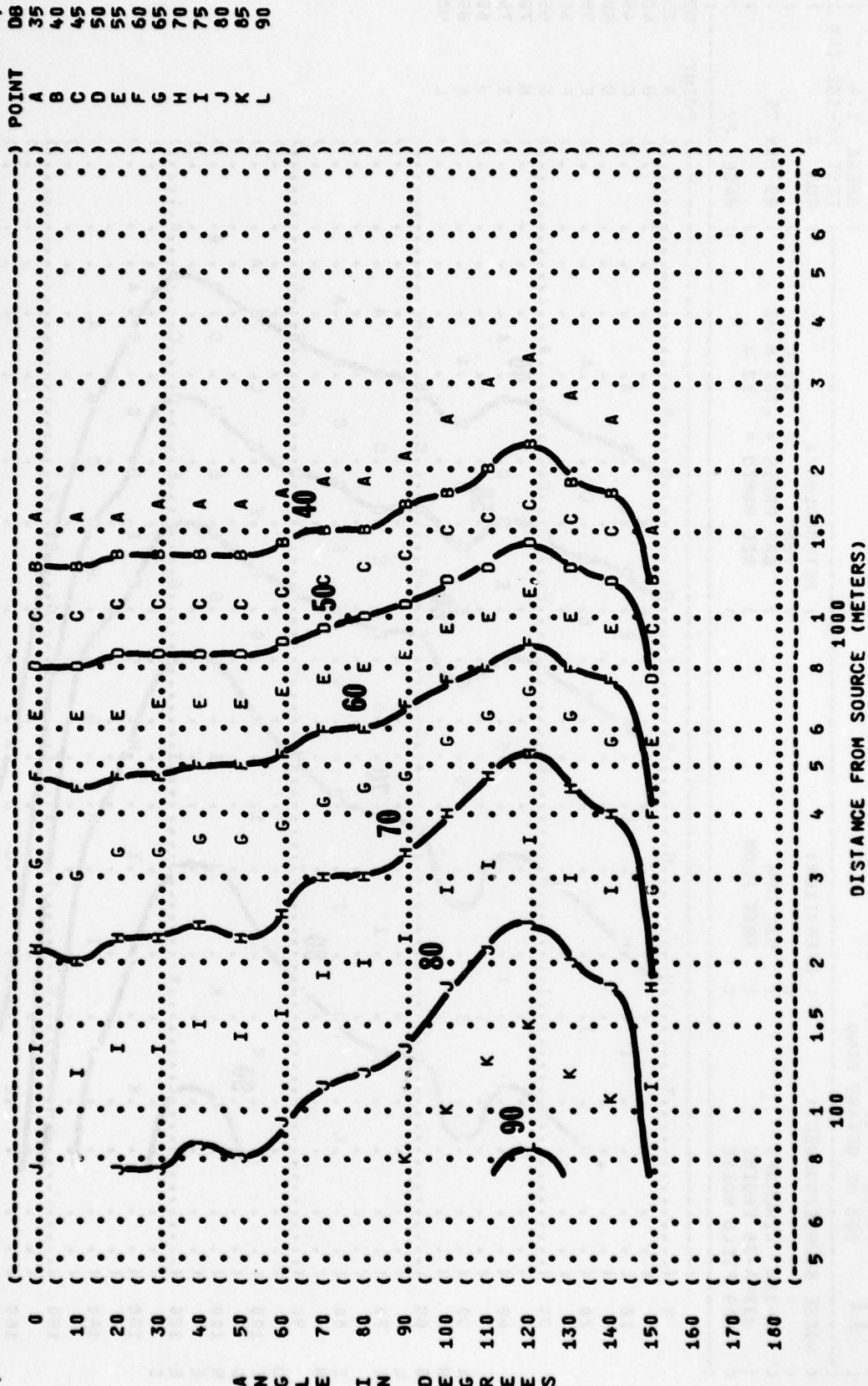


(FIGURE: SOUND PRESSURE LEVEL (SPL))
 (11 EQUAL LEVEL CONTOURS (DB))
 (63 HZ OCTAVE BAND)
 (NOISE SOURCE/SUBJECT:)
 (T-33A AIRCRAFT)
 (J33-A-35 ENGINE)
 (FAR FIELD NOISE)
 (OPERATION:)
 (50% RPM)
 (FREE FLOW)
 (METEOROLOGY:)
 (TEMP = 15 C)
 (BAR PRESS = .760 M HG)
 (REL HUMID = 70 %)
 (IDENTIFICATION:)
 (OMEGA 1.4)
 (TEST 75-002-045)
 (RUN 02)
 (09 MAY 75)
 (PAGE 19)



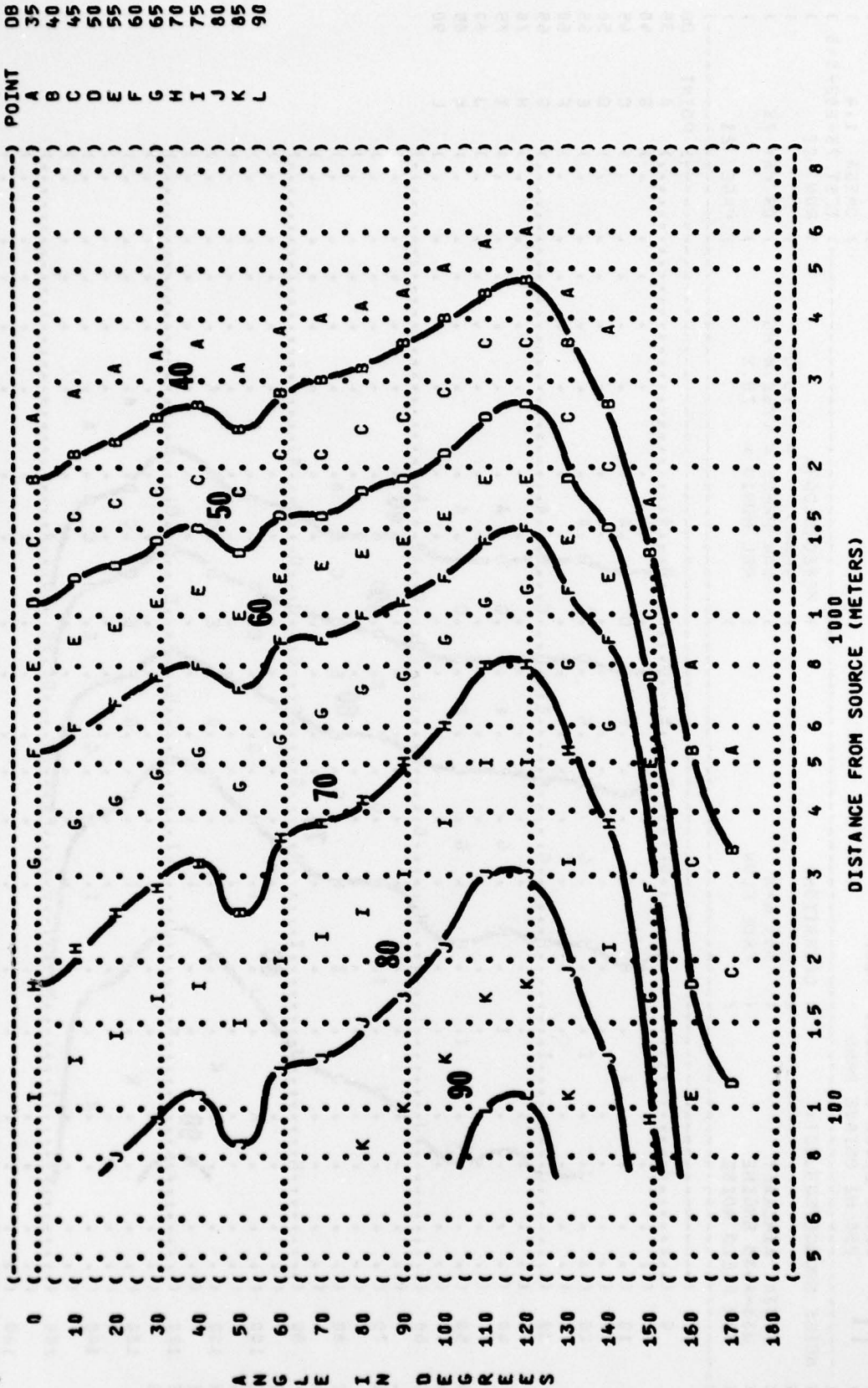
A N G L E I N D E G R E E S

(FIGURE: SOUND PRESSURE LEVEL (SPL))
 (11 EQUAL LEVEL CONTOURS (DB))
 (250 HZ OCTAVE BAND)
 (NOISE SOURCE/SUBJECT:)
 (T-33A AIRCRAFT)
 (J33-A-35 ENGINE)
 (FAR FIELD NOISE)
 (OPERATION:)
 (50% RPM)
 (FREE FLOW)
 (METEOROLOGY:)
 (TEMP = 15 C)
 (BAR PRESS = .760 M HG)
 (REL HUMID = 70 %)
 (IDENTIFICATION:)
 (OMEGA 1.4)
 (TEST 75-002-045)
 (RUN 02)
 (09 MAY 75)
 (PAGE 21)



A N G L E I N D E G R E E S

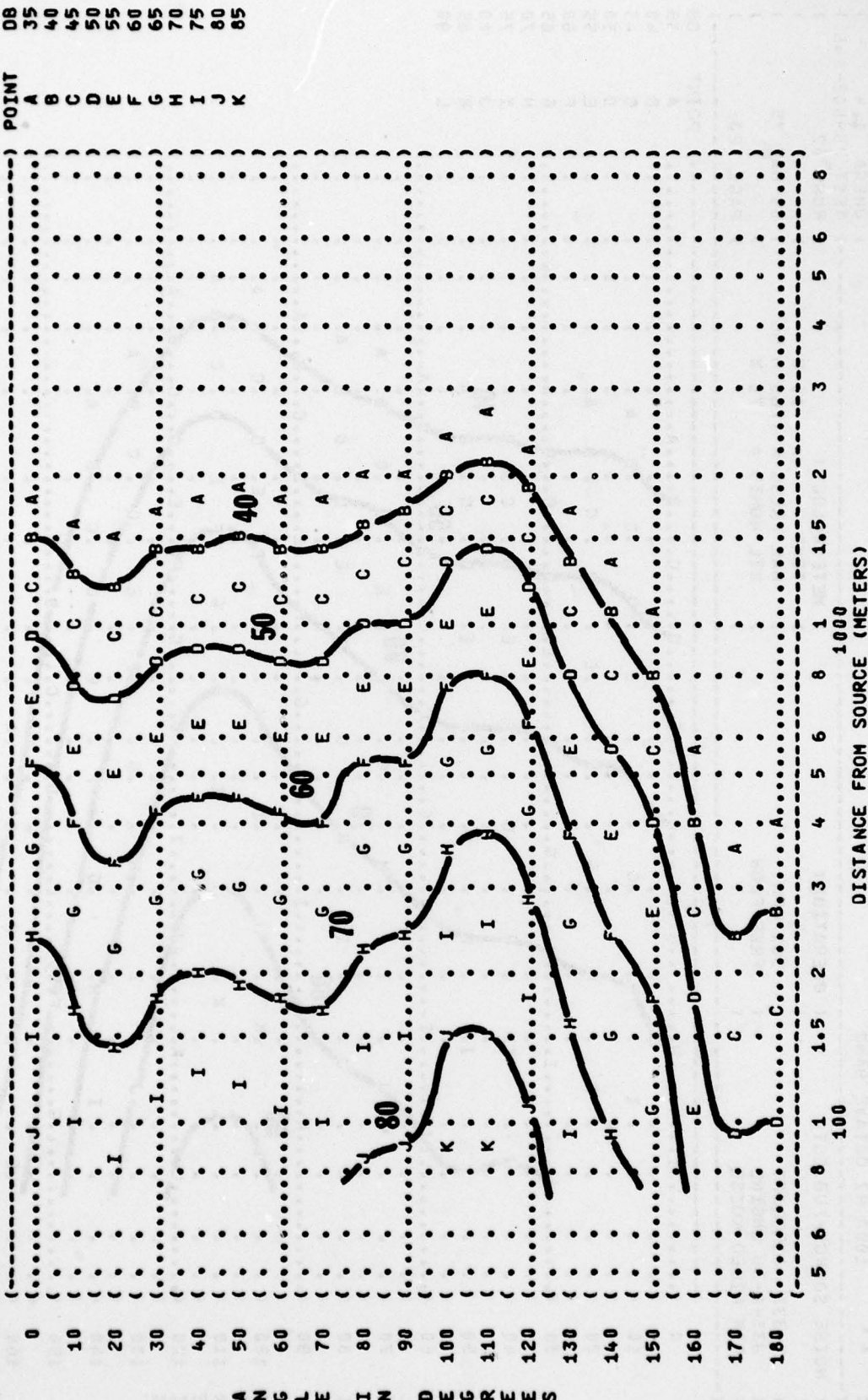
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 (11)
 (500 HZ OCTAVE BAND)
 (NOISE SOURCE/SUBJECT:)
 (T-33A AIRCRAFT)
 (J33-A-35 ENGINE)
 (FAR FIELD NOISE)
 (OPERATION:)
 (50% RPM)
 (FREE FLOW)
 (METEOROLOGY:)
 (TEMP = 15 C)
 (BAR PRESS = .760 M HG)
 (REL HUMID = 70 %)
 (IDENTIFICATION:)
 (OMEGA 1.4)
 (TEST 75-002-045)
 (RUN 02)
 (09 MAY 75)
 (PAGE 22)





100

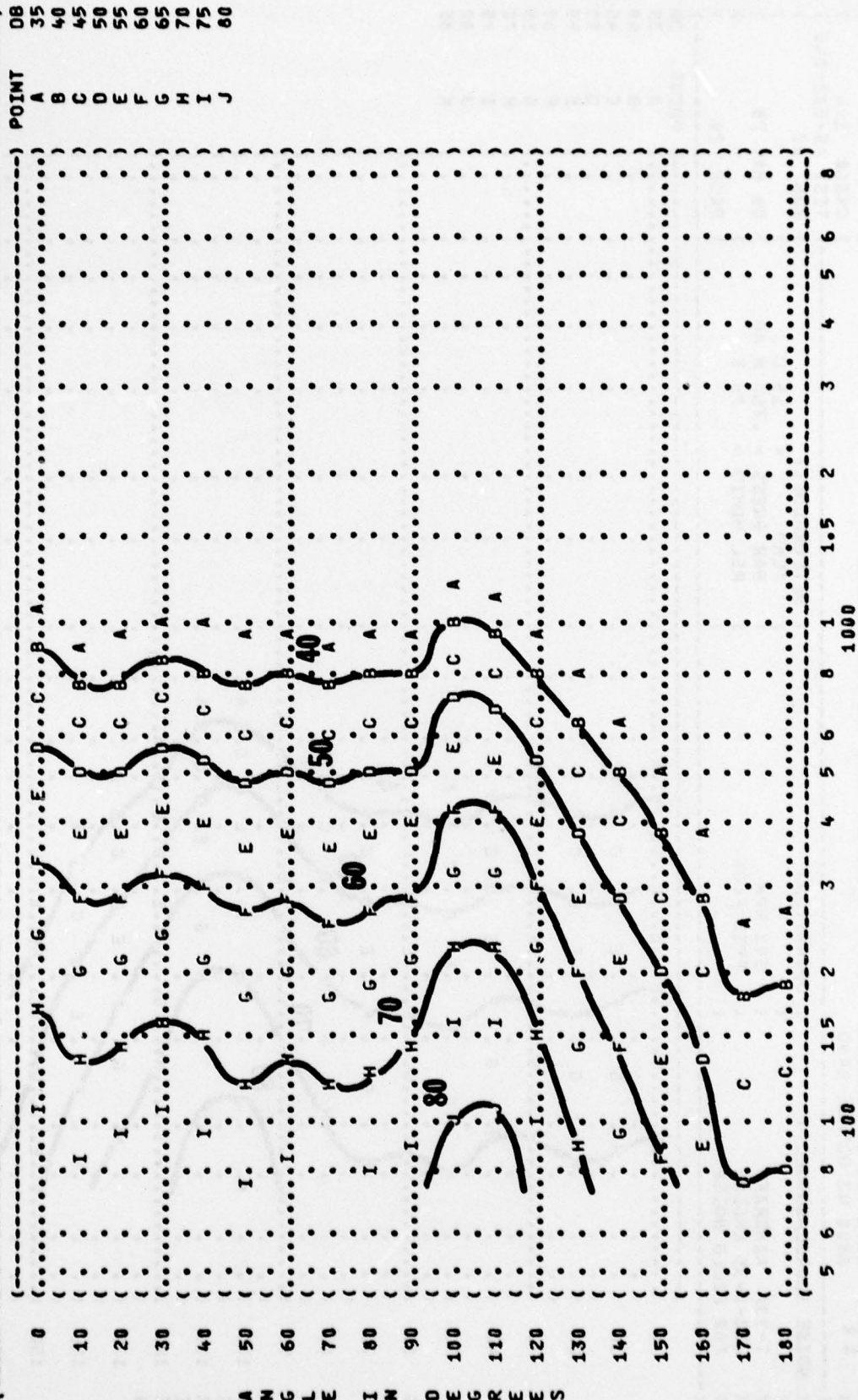
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 (EQUAL LEVEL CONTOURS (DB)
 (11 2000 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION:
 (T-33A AIRCRAFT (50% RPM
 (J33-A-35 ENGINE (FREE FLOW
 (FAR FIELD NOISE ()
 () METEOROLOGY:
 () TEMP = 15 C
 () BAR PRESS = .760 M HG
 () REL HUMID = 70 %
 () OMEGA 1.4
 () TEST 75-002-045
 () RUN 02
 () 09 MAY 75
 () PAGE 24



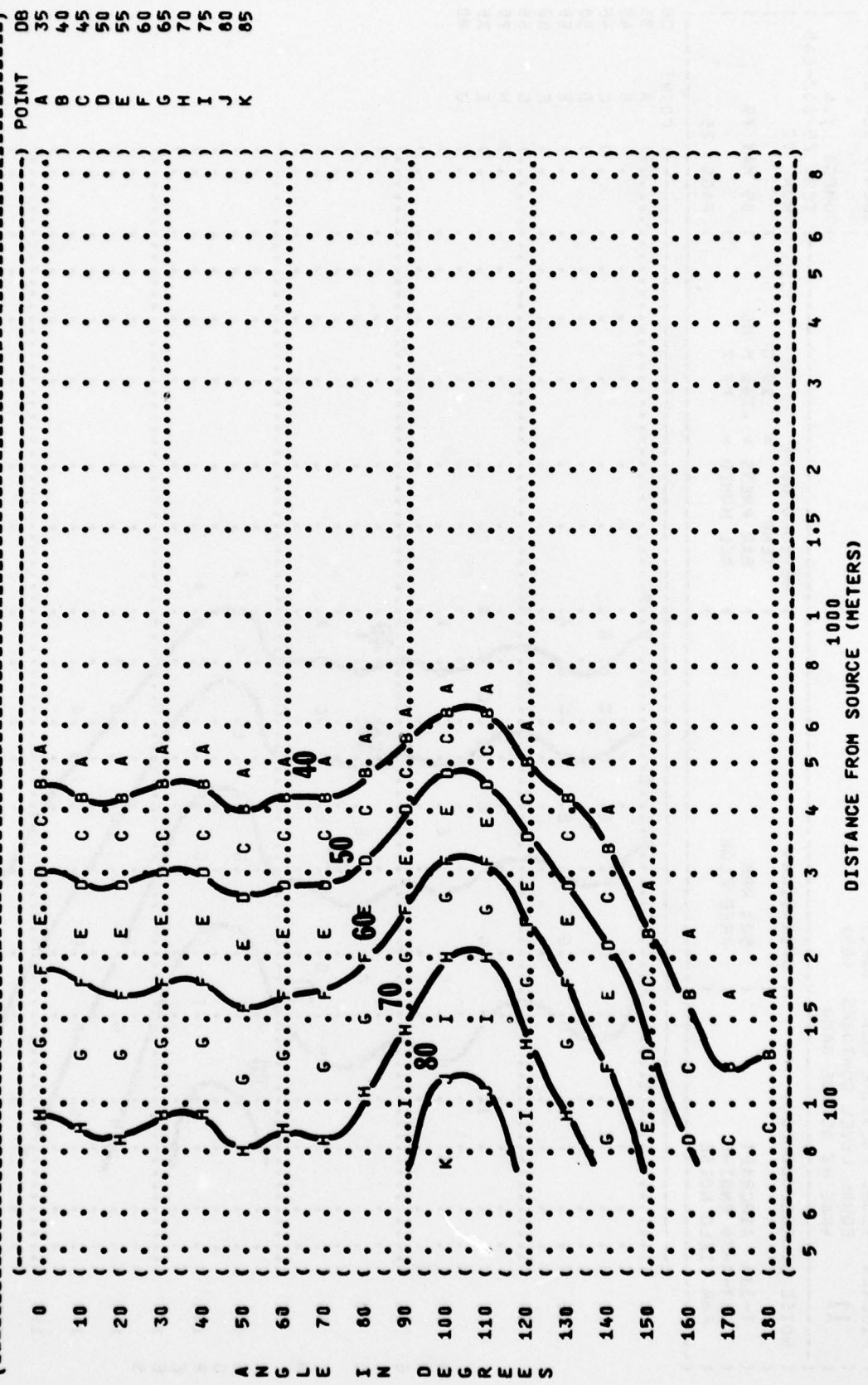
DISTANCE FROM SOURCE (METERS)

A N G L E I N D E G R E E S

(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (11 EQUAL LEVEL CONTOURS (DB)
 (4000 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION:
 (T-33A AIRCRAFT (50% RPM
 (J33-A-35 ENGINE (FREE FLOW
 (FAR FIELD NOISE ()
 () METEOROLOGY: ()
 () TEMP = 15 C
 () BAR PRESS = .760 M HG
 () REL HUMID = 70 %
 () PAGE 25
 () IDENTIFICATION:
 () OMEGA 1.4
 () TEST 75-002-045
 () RUN 02
 () 09 MAY 75

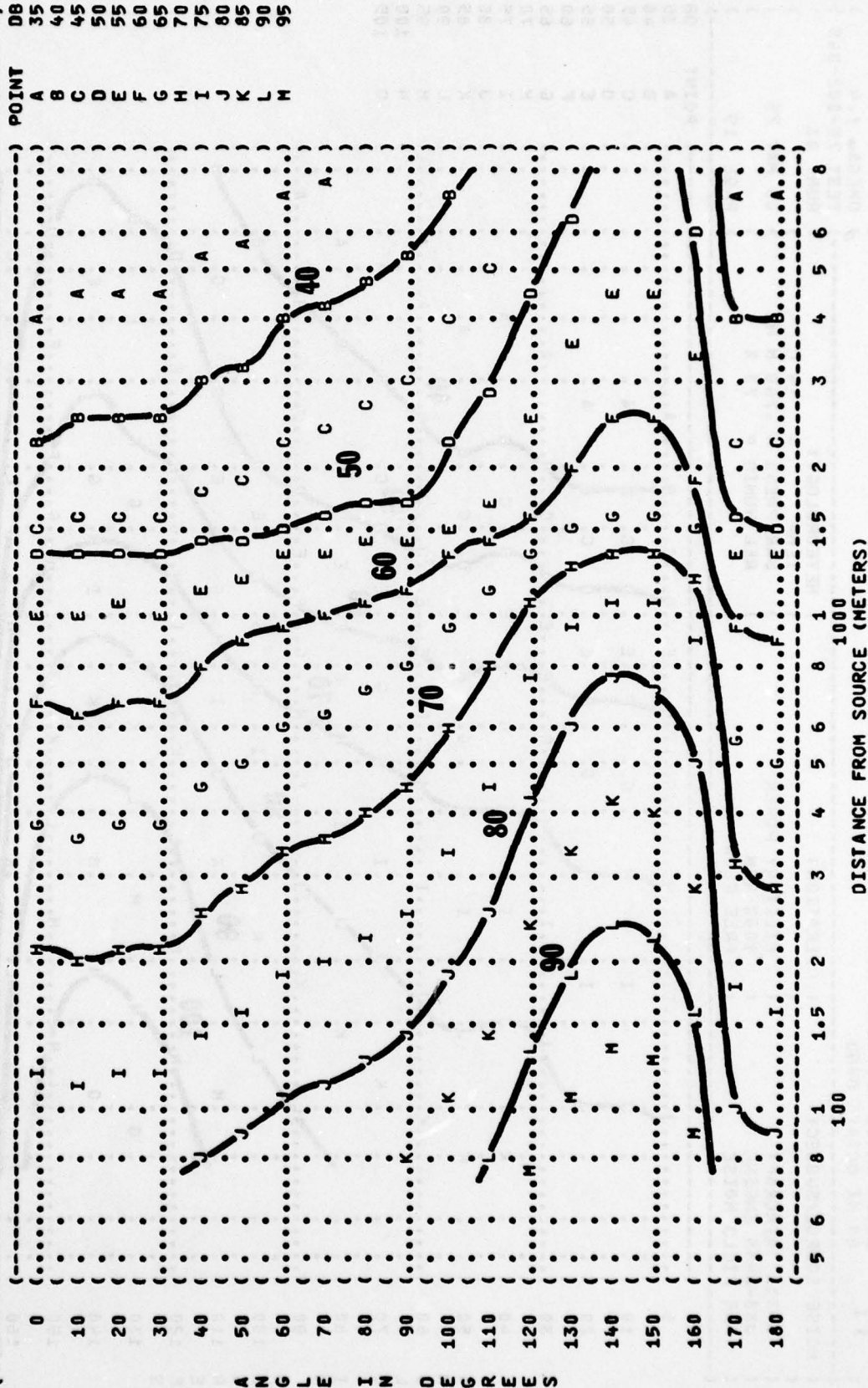


() FIGURE: SOUND PRESSURE LEVEL (SPL)
 () EQUAL LEVEL CONTOURS (DB)
 () 11 8000 HZ OCTAVE BAND
 () NOISE SOURCE/SUBJECT: () OPERATION:
 () () 50% RPM
 () J33-A-35 ENGINE () FREE FLOW
 () FAR FIELD NOISE ()
 () METEOROLOGY: () TEMP = 15 C
 () BAR PRESS = .760 M HG
 () REL HUMID = 70 %
 () PAGE 26
 () IDENTIFICATION: ()
 () OMEGA 1.4
 () TEST 75-002-045
 () RUN 02

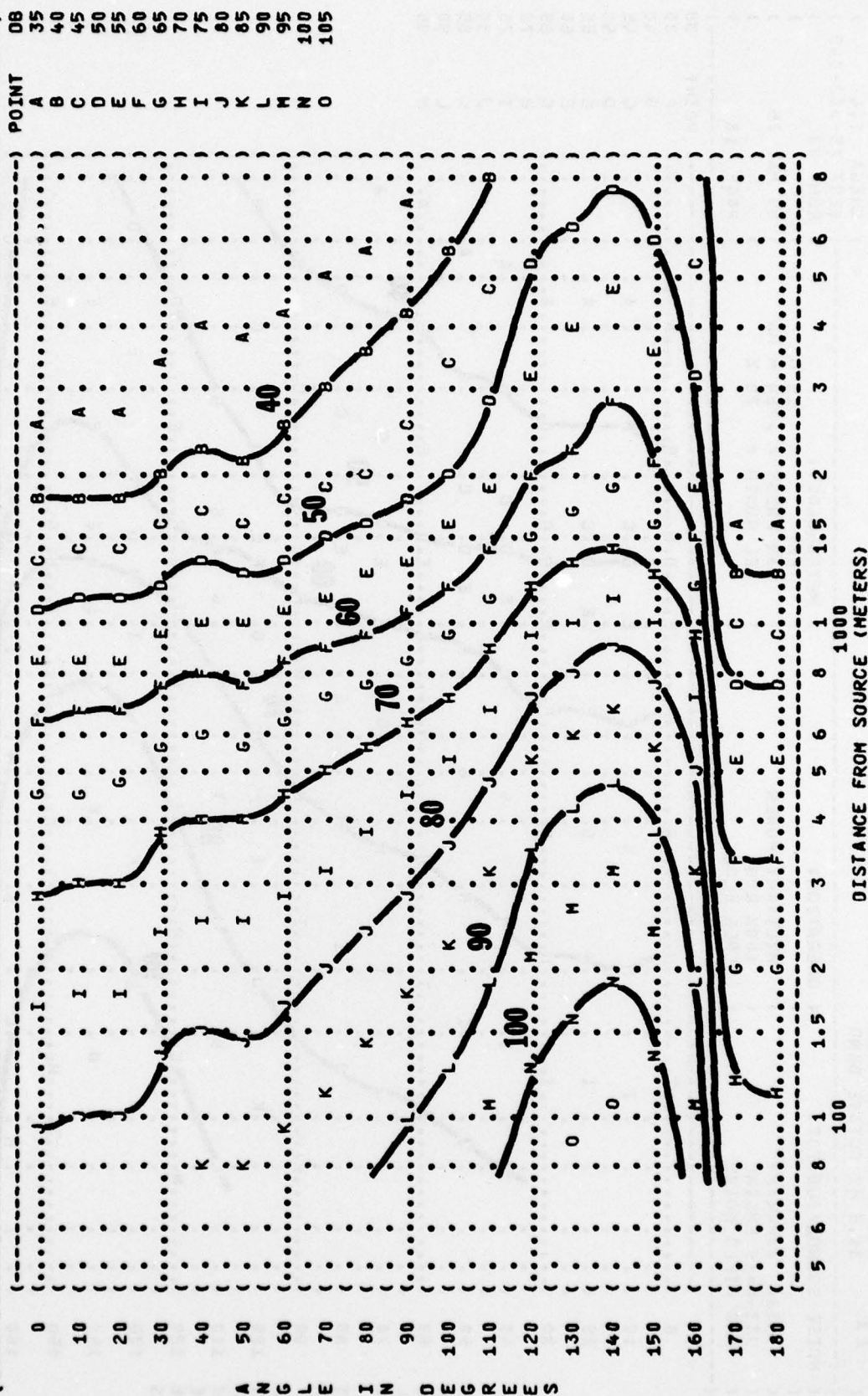


A N G L E I N D E G R E E S

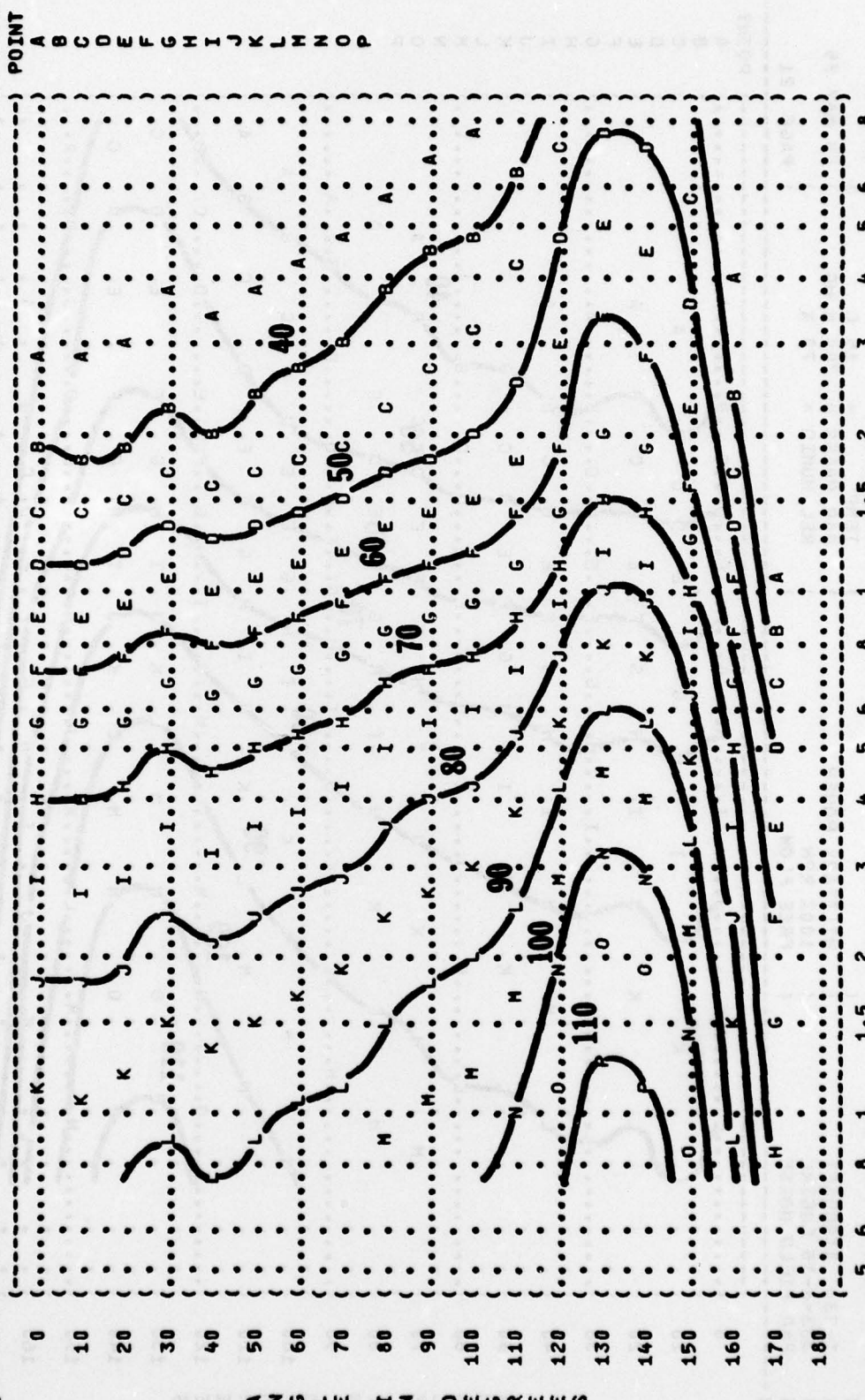
(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (11 EQUAL LEVEL CONTOURS (DB)
 (31.5 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION:
 (T-33A AIRCRAFT (MILITARY POWER
 (J33-A-35 ENGINE (100% RPM
 (FAR FIELD NOISE (FREE FLOW
 (METEOROLOGY:
 (TEMP = 15 C
 (BAR PRESS = .760 M HG
 (REL HUMID = 70 %
 (IDENTIFICATION:
 (OMEGA 1.4
 (TEST 75-002-045
 (RUN 03
 (09 MAY 75
 (PAGE 10



() FIGURE: SOUND PRESSURE LEVEL (SPL)
 () 11 EQUAL LEVEL CONTOURS (DB)
 () 63 HZ OCTAVE BAND
 () NOISE SOURCE/SUBJECT:
 () T-33A AIRCRAFT
 () J33-A-35 ENGINE
 () FAR FIELD NOISE
 () OPERATION:
 () MILITARY POWER
 () 100% RPM
 () FREE FLOW
 () METEOROLOGY:
 () TEMP = 15 C
 () BAR PRESS = .760 M HG
 () REL HUMID = 70 %
 () IDENTIFICATION:
 () OMEGA 1.4
 () TEST 75-002-045
 () RUN 03
 () 09 MAY 75
 () PAGE 19



IDENTIFICATION:
OMEGA 1.4
TEST 75-002-045
RUN 03
09 MAY 75
PAGE 20



((FIGURE: SOUND PRESSURE LEVEL (SPL)
 ((EQUAL LEVEL CONTOURS (DB)
 ((11 250 HZ OCTAVE BAND
 ((NOISE SOURCE/SUBJECT: (OPERATION:
 ((T-33A AIRCRAFT (MILITARY POWER
 ((J33-A-35 ENGINE (100% RPM
 ((FAR FIELD NOISE (FREE FLOW
 ((METEOROLOGY: (TEMP = 15 C
 ((BAR PRESS = .760 M HG
 ((REL HUMID = 70 %
 ((IDENTIFICATION: (OMEGA 1.4
 ((TEST 75-002-045
 ((RUN 03
 ((09 MAY 75
 ((PAGE 21

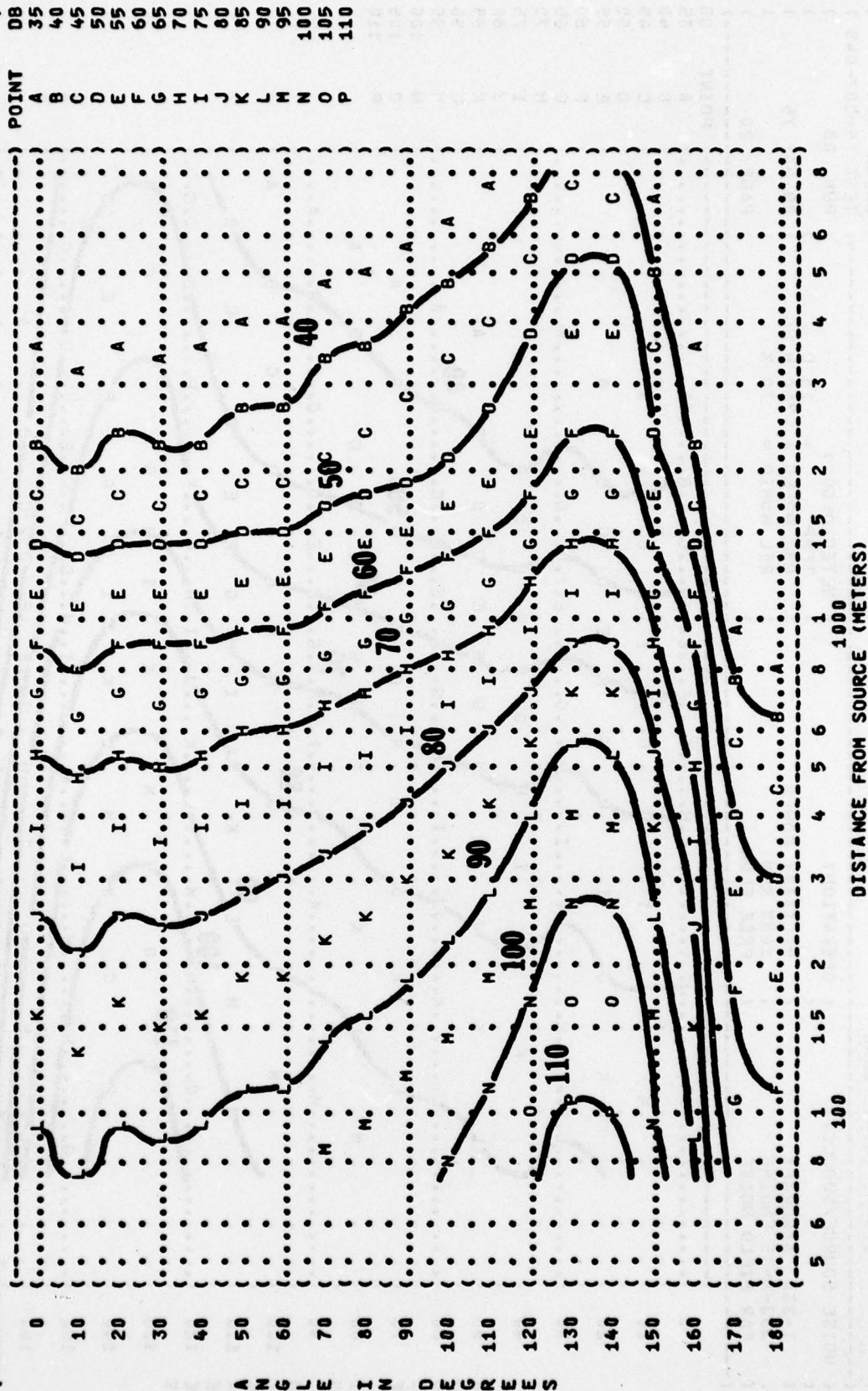


FIGURE: SOUND PRESSURE LEVEL (SPL)
EQUAL LEVEL CONTOURS (DB)
500 HZ OCTAVE BAND

11

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT
J33-A-35 ENGINE
FAR FIELD NOISE

OPERATION:

MILITARY POWER
100% RPM
FREE FLOW

METEOROLOGY:

TEMP = 15 C
BAR PRESS = .760 MM HG
REL HUMID = 70 %

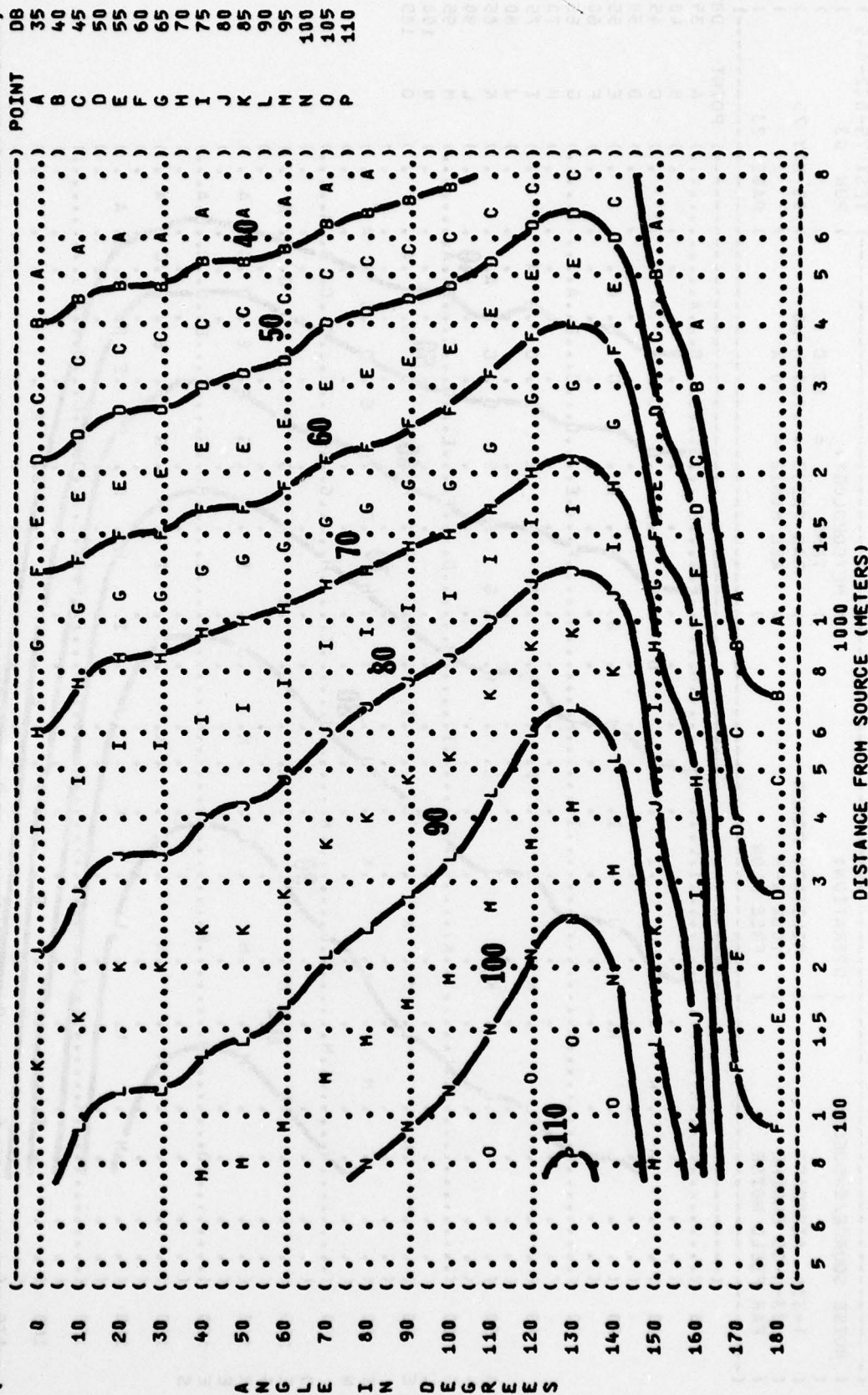
OMEGA 1.4

TEST 75-002-045

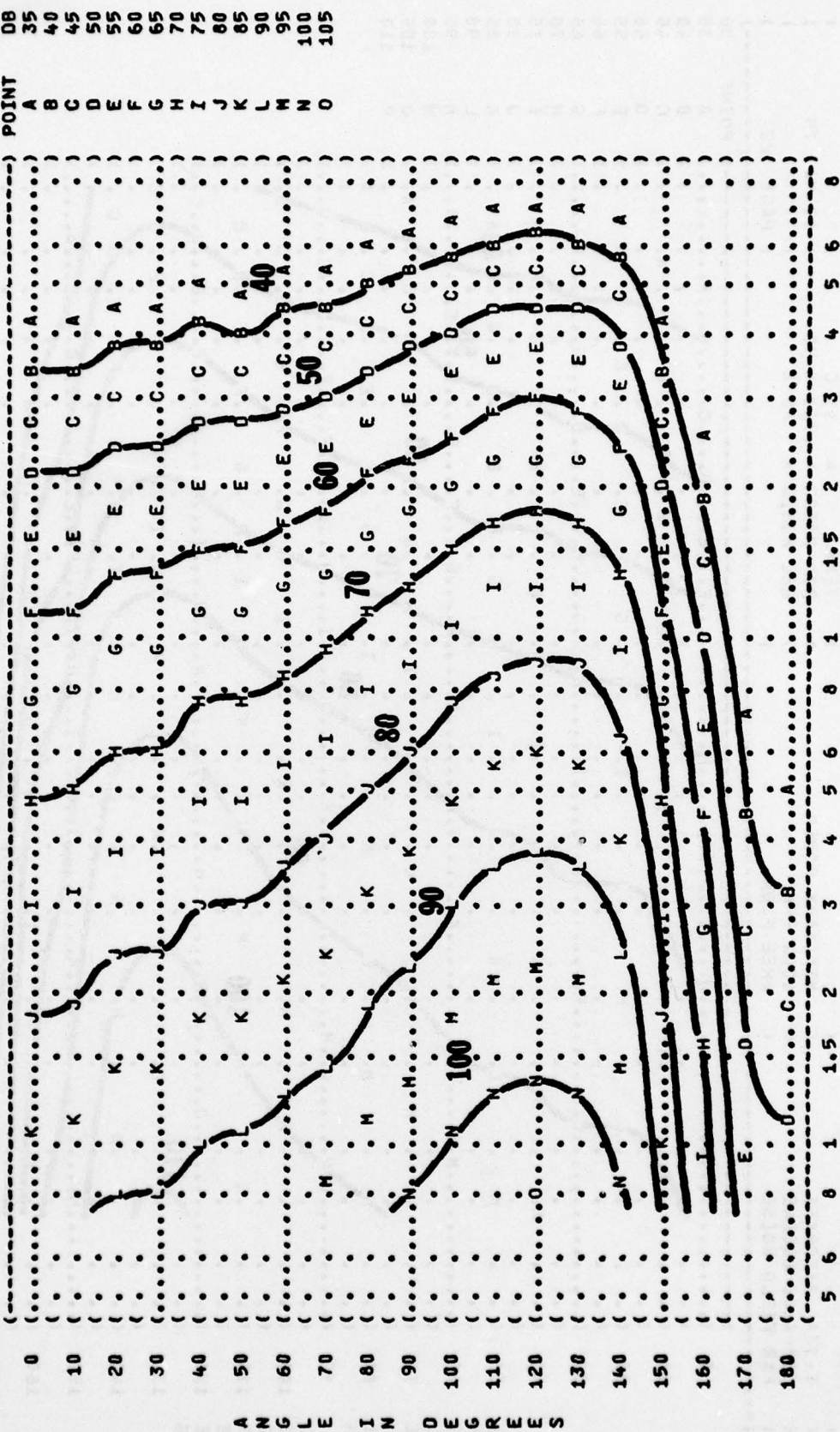
RUN 03

09 MAY 75

PAGE 22



(FIGURE: SOUND PRESSURE LEVEL (SPL))
 (11 EQUAL LEVEL CONTOURS (DB))
 (NOISE SOURCE/SUBJECT:)
 (T-33A AIRCRAFT)
 (J33-A-35 ENGINE)
 (FAR FIELD NOISE)
 (OPERATION:)
 (MILITARY POWER)
 (100% RPM)
 (FREE FLOW)
 (METEOROLOGY:)
 (TEMP = 15 C)
 (BAR PRESS = .760 M HG)
 (REL HUMID = 70 %)
 (IDENTIFICATION:)
 (OMEGA 1.4)
 (TEST 75-002-045)
 (RUN 03)
 (09 MAY 75)
 (PAGE 23)



DISTANCE FROM SOURCE (METERS)

(FIGURE: SOUND PRESSURE LEVEL (SPL)
 (EQUAL LEVEL CONTOURS (DB)
 (11 2000 HZ OCTAVE BAND
 (NOISE SOURCE/SUBJECT: (OPERATION:
 (T-33A AIRCRAFT (MILITARY POWER
 (J33-A-35 ENGINE (100% RPM
 (FIELD NOISE (FREE FLOW
 (METEOROLOGY: TEMP = 15 C
 (BAR PRESS = .760 M HG
 (REL HUMID = 70 %
 (RUN 03
 (09 MAY 75
 (PAGE 24
 (IDENTIFICATION:
 (OMEGA 1.4
 (TEST 75-002-045
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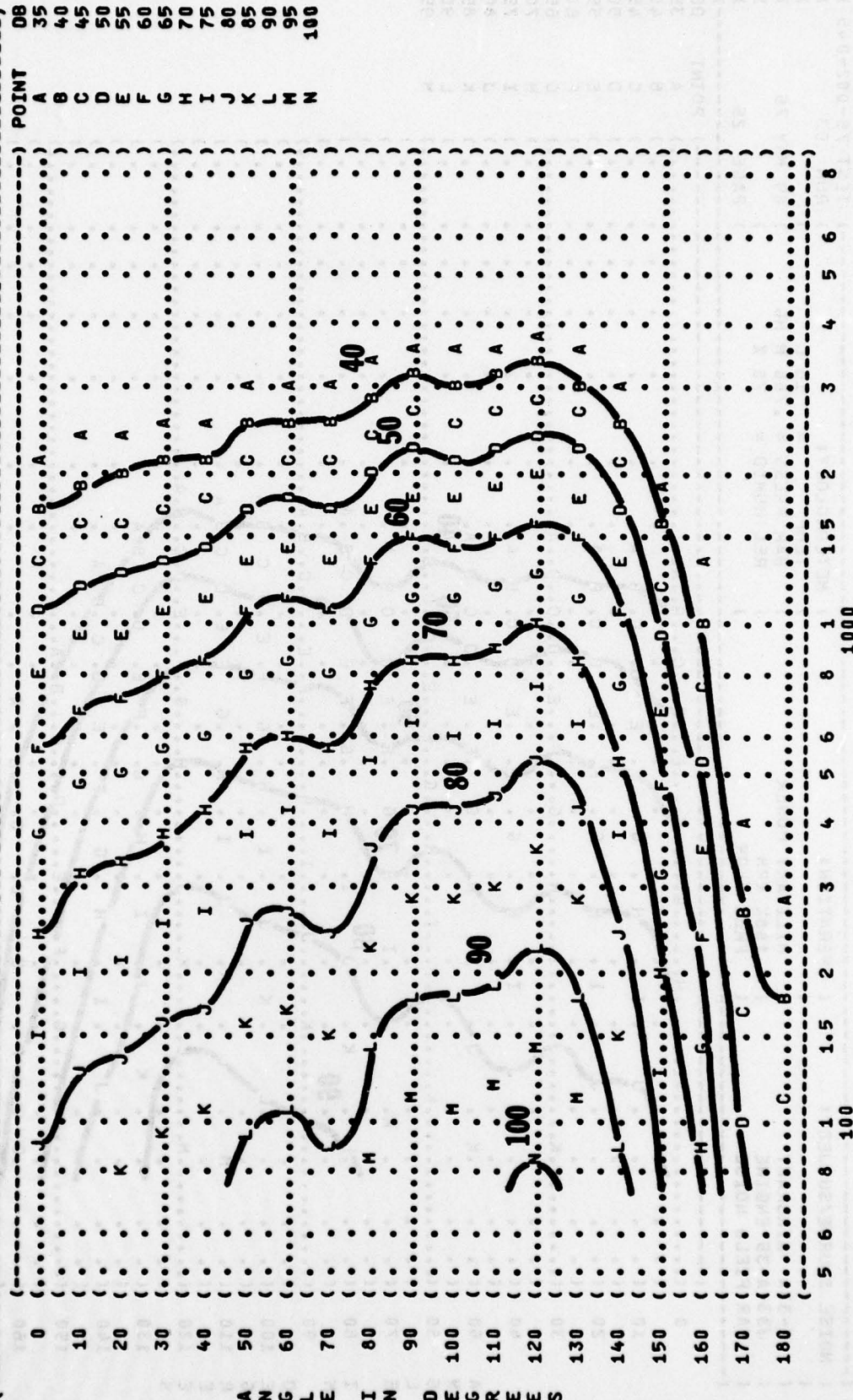


FIGURE: SOUND PRESSURE LEVEL (SPL)
 11 EQUAL LEVEL CONTOURS (DB)
 6000 HZ OCTAVE BAND
 NOISE SOURCE/SUBJECT: (OPERATION:)
 T-33A AIRCRAFT (MILITARY POWER)
 J33-A-35 ENGINE (100% RPM)
 FAR FIELD NOISE (FREE FLOW)
 METEOROLOGY: ()
 TEMP = 15 C
 BAR PRESS = .760 M HG
 REL HUMID = 70 %
 IDENTIFICATION: ()
 OMEGA 1.4
 TEST 75-002-045
 RUN 03
 09 MAY 75
 PAGE 26

